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Experimental results for structural design of Wire-and-Arc Additive Manufactured stainless steel members

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This is the final peer-reviewed accepted manuscript of:

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# 1 **Experimental results for structural design of Wire-and-Arc Additive** 2 **Manufactured stainless steel members**

3  
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## 9 10 **Abstract**

11 Additive Manufacturing has gained great importance in the recent development to produce metallic  
12 structural elements for civil engineering applications. While a lot of research effort has been focused  
13 on different technologies (such as Powder Bed Fusion), there is still quite limited knowledge  
14 concerning the structural response of Wire-and-Arc Additive Manufactured (WAAM) metallic  
15 elements, as very few experimental campaigns aimed at assessing their geometrical and mechanical  
16 properties have been carried out. The paper presents selected results of a wide experimental campaign  
17 focused on the assessment of the main geometrical and mechanical properties of Wire-and-Arc  
18 Additive Manufactured (WAAM) stainless steel material, carried out at the Topography and  
19 Structural Engineering Labs from University of Bologna. In detail, the focus of the present study is  
20 on the characterization of the surface irregularities by means of various measuring techniques and on  
21 the main mechanical properties of the material, in terms of tensile and compressive strengths,  
22 Young's modulus and post elastic behavior. Tests results have been interpreted through statistical  
23 tools in order to derive mean values and gather information about the variability of both geometrical  
24 and mechanical parameters.

## 25 26 **Key words**

27 Additive manufacturing; Experiment; Wire-and-arc; Stainless steel; Testing.

## 1           **1. Introduction**

2   Along the centuries, the evolution in building construction has always been strictly linked to  
3   significant advancements in material science, technology, industrial processes and engineering. As  
4   the first Industrial Revolution marked the use of metals as structural members, the improvement in  
5   scientific knowledge and the introduction of concrete at the beginning of XX century paved the way  
6   for Reinforced Concrete (RC) constructions ([1] Addis 2007). Similarly, in the last 30 years the way  
7   structures are designed, planned and built has changed completely with the technological innovations  
8   offered by software for Computer-Aided Design (CAD), which replaced manual drawings, without  
9   however changing the resulting architectural shapes. Only in the early years of the 21<sup>st</sup> century, a new  
10   design approach to architecture resulting in novel forms started to emerge by the utilization of three-  
11   dimensional computer modelling and digital fabrication methods. More recently, with what has been  
12   called the *digital turn* ([2] Carpo 2013), innovative computerized tools for architecture, structural and  
13   civil engineering have gained influence, enabling the design and construction of buildings with  
14   complex, doubly-curved geometry, such as shell structures and other free-form designs ([3]  
15   Adriaenssens et al. 2014).

16   The growth of automation since the beginning of 21<sup>st</sup> century has prevailed in almost all production  
17   domains with the exception of the building construction sector, in which the use of automatic tools is  
18   still challenging and at its first applications. The main challenge is due to some peculiar aspects of  
19   the construction industry: (i) building and civil constructions are very large-scale products requiring  
20   customization of conventional automated fabrication technologies; (ii) conventional design  
21   approaches are not tailored for automation; (iii) significantly smaller ratio of production quantity/type  
22   of final products as compared with other industries; (iv) limitations in the materials that could be  
23   employed by an automated system ([4] Koshnevis 2004). Only in the last few years, the fast  
24   development in digital fabrication techniques is leading towards applications in Structural  
25   Engineering field as well, through Additive Manufacturing (AM)-based technologies, already  
26   commonly used in other sectors such as aerospace, automotive and biomedical engineering ([5]  
27   Attaran 2017, [6] Thomas et al. 1996, [7] Song et al. 2002, [8] Giannatsis et al. 2009).

28   The term Additive Manufacturing has been attempted to be standardized in ISO/ASTM 52900 [9],  
29   where such terminology has been adopted for all processes of making parts from 3D models and  
30   materials. Nowadays Additive Manufacturing (AM) refers to the technology (or additive process) of  
31   depositing successive thin layers of material upon each other, as opposed to the traditional subtractive  
32   manufacturing, producing a final three-dimensional object, realized either in plastic, resin, rubber,  
33   ceramic, glass, concrete and metal. Recent reviews of the metal Additive Manufacturing processes

1 ([10] Wong and Hernandez 2012, [11] Levandowski and Seifi 2016, [12] Sames et al 2016, [13]  
2 Everton et al. 2016, [14] Buchanan and Gardner 2019) distinguish them into three major categories:  
3 (i) Powder Bed Fusion (PBF); (ii) Directed Energy Deposition (DED) and (iii) sheet lamination.  
4 In recent years research effort has been intensively dedicated to the study of PBF material fabrication  
5 and its specifications concerning building orientation and mechanical characteristics ([15] Skiba et  
6 al. 2009, [16] Niendorf et al. 2013, [17] Guan et al. 2013, [18] Yap et al. 2015; [19] Song et al. 2015,  
7 [20] Buchanan et al. 2017). ASTM F2792-10 ([21]) provides also an outline of metallic single-step  
8 Additive Manufacturing methods for this category of AM process. The PBF process, although  
9 presenting small geometrical imperfections and mechanical characteristics in the order of traditional  
10 stainless steel material, is limited by the build envelope of the equipment, which is typically a 250-  
11 mm cube ([20] Buchanan et al. 2017).

12 On the other hand, DED processes, and in particular Wire-and-Arc Additive Manufacturing (WAAM)  
13 technology, lies upon a more flexible building set-up, which consists of a “printing head” at the top  
14 of a robotic arm, allowing for the realization of elements without theoretical dimensional constraints.  
15 For such reason, it appears more suitable for structural engineering applications, for which the outputs  
16 requested are of the order of several meters (typically 3 to 5 m long). However, in order to obtain  
17 pieces of large dimensions, higher printing velocities are required, resulting in larger geometrical  
18 imperfections with respect to the digital model. Therefore, much effort is needed for a proper  
19 assessment of both the geometrical and mechanical characterization of the outputs from Wire-and-  
20 Arc Additive Manufacturing (WAAM) process.

## 2. Wire-and-Arc Additive Manufacturing for structural applications

### 2.1 The MX3D manufacturing process

23 A basic AM system consists of a combination of a motion system, heat source and feedstock ([10]  
24 Wong and Hernandez 2012). In particular, the combination of an electric arc as heat source and wire  
25 as feedstock is referred to as Wire-and-Arc Additive Manufacturing (WAAM), which currently uses  
26 standard off-the-shelf welding equipment, such as welding power source, torches and wire feeding  
27 system, while motion is provided by either robotic systems or computer numerical-controlled  
28 gantries. WAAM’s layer height is commonly in the range of 1 to 2 mm, resulting in an expected  
29 surface roughness of about 0.5 mm for single track deposits. As a result, this process is not considered  
30 net shape, as machining is required to finish the part, thus being better suited for low- to medium-  
31 complexity and medium- to large-scale elements, as those implemented in Structural Engineering  
32 ([22] Williams et al. 2016, [23] Uziel 2016, [24] Haden et al. 2017, [25] Ji et al. 2017).

1 Among the various metal Additive Manufacturing technologies, the present work focuses on the  
2 Wire-and-Arc Additive Manufacturing (WAAM) process developed by the Dutch company MX3D  
3 ([26] [www.mx3d.com](http://www.mx3d.com)) and adopted to realize the first 3D-printed steel footbridge (Figure 1) which  
4 will be placed in the city center of Amsterdam by 2020 ([27] Joosten 2015, [28] van Bolderen 2017).  
5 Indeed, the authors are partner in the “Bridge project” held by MX3D to run experimental tests for  
6 the characterization of small specimens taken from planar sheets and full-scaled tubes.

7 MX3D makes use of gas metal arc welding, a process characterized by a continuous wire electrode  
8 which is drawn from a reel by an automatic wire feeder, through the contact tip in the welding torch.  
9 The heat is transferred from the welding arc and the internal resistive power causes the wire to melt  
10 ([29] Kim et al. 2003, [30] Yilmax and Uglu 2017).

11 The motion system used by MX3D consists of industrial multi-axis ABB robots which, theoretically,  
12 are able to print from any angle. Two different printing strategies have been explored by MX3D: a  
13 so-called *continuous printing*, meaning that the material is deposited in continuous, and a so-called  
14 *dot-by-dot printing*, meaning that the material is deposited by successive points. The effects of these  
15 strategies on the metallurgic characteristics have been analyzed in [28]. The process used to realize  
16 the MX3D Bridge [26] is a *continuous* printing strategy, and the welding wire used for the structure  
17 under study is 308LSi austenitic stainless steel.

18 The fundamental process parameters are (i) the current and its voltage, (ii) the wire diameter, (iii) the  
19 wire-feed rate, (iv) the welding speed and (v) the vertical printed layer height. The combination of  
20 such controlling parameters affects the printing quality (geometrical precision and surface roughness)  
21 as well as the material mechanical properties. Specifications on the parameters adopted in the  
22 realization of the MX3D Bridge [26] are the following: 1-mm wire diameter, 0.6 to 2 mm/min wire-  
23 feed rate, 15 to 30 mm/sec welding speed, 1 to 3 mm vertical printed layer height and an average total  
24 printing speed of 0.5 to 2 kg/hour.

25 For Structural Engineering applications, the need of high welding velocity for a rapid realization of  
26 structural elements of such proportions plays a crucial role for the specific characteristics of the  
27 printed parts, as it induces geometric inaccuracy of the outcomes, both in terms of surface roughness  
28 and lack of straightness of the elements. For a given element to be printed, a digital model from which  
29 the printing head reads the coordinates of the points defining step by step the position of the welded  
30 layer is created with Rhinoceros software ([31] <http://rhino3d.com/>). However, due to intrinsic  
31 inaccuracy of the printing process, each point of the digital model has a real counterpart whose  
32 position is not exactly the one of the digital model, as it is affected by an error.



1  
2 *Figure 1: MX3D footbridge realized with WAM process and presented at the Dutch Design Week 2018 in Eindhoven [26].*  
3

## 4 **2.2 Objectives of the work**

5 The specifications of the innovative manufacturing process are crucial to derive the response in terms  
6 of structural performances of the printed elements. The large velocity of deposition used to realize  
7 the specimens has induced some geometric irregularities which should be properly taken into account  
8 both in the design and construction phases. Moreover, the heating process induces some non-  
9 negligible residual stresses and deformation of the crystalized structure of the stainless steel, which  
10 may lead to an orthotropic behavior and should be further analyzed as well. For instance, Figure 2  
11 provides different views of typical tubular elements manufactured by MX3D using the same  
12 continuous printing strategy as the one adopted for the realization of the MX3D Bridge. The photos  
13 allow to visually appreciate the typical geometrical imperfections resulting from the manufacturing  
14 process: cross-section shape irregularity (Figure 2a), lack of straightness (Figure 2b) and surface  
15 roughness (Figure 2c). These geometrical imperfections together with the specific material  
16 mechanical behavior related to the printing parameters affect the structural response of the printed  
17 elements and have to be fully addressed in order to provide valuable information to be used in the  
18 structural design.

19 In the present work the attention is focused on the geometrical and mechanical characterization of  
20 samples (planar specimens and hollow tubes) manufactured by MX3D using the Wire-and-Arc  
21 Additive Manufacturing process that has been used, for the first time in a civil engineering context,  
22 for the MX3D Bridge [26].



Figure 2: Geometric issues of WAAM tubular elements: (a) irregular cross-section; (b) lack of straightness; (c) surface roughness.

### 3. Geometrical and mechanical parameters of WAAM elements

In the present Section, meaningful definitions and quantities adopted to describe the geometrical and mechanical properties of the specimens and corresponding digital models are introduced. All quantities describing the geometry of the digital model will be hereafter indicated with the suffix “*n*”, standing for “*nominal*”, while those corresponding to the real printed element will be indicated with the suffix “*real*”.

Two different categories of specimens are studied: rectangular plates of constant thickness (Figure 3) from which “dog-bone” shaped specimens have been obtained, and both hollow short (also referred to as stub columns) and long circular tubes of equal cross-section and various lengths (Figure 4).

#### 3.1 Geometrical parameters

Considering the uniform rectangular plate represented in Figure 3, the origin of coordinate system used to describe the geometry of the plate is assumed to be located at one edge. The  $x$  and  $y$  axes are taken parallel to the two main directions of the plate, while the  $z$ -axis is perpendicular to the  $x$ - $y$  plane. The thickness of the plate is given by the amount of welded metal positioned by the printing head, whose nominal value in the digital model is constant all over the plate and equal to  $t_n$ . On the other hand, the value of thickness in the printed plate is in general non constant and varies both with  $x$  and  $y$ , so that  $t_{real}=t_{real}(x,y)$ . Given that the plates have been produced with a certain printing direction, the surface roughness might have a different influence on the mechanical behavior of specimens cut along the printing direction ( $x$ ) with respect to those cut perpendicular to it ( $y$ ).



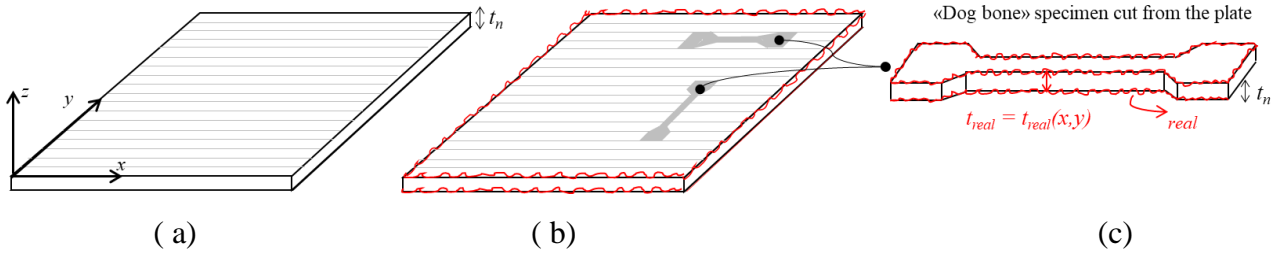


Figure 3: Digital model (a) and corresponding printed element (b) of a WAAM plate from which "dog-bone" specimens (c) are extracted.

Regarding the uniform hollow cylinder represented in Figure 4, a cylindrical coordinate system  $(r, z, \theta)$  is adopted to describe the geometry of the tubular element, differentiating between the parameters used as inputs for the digital model (nominal values) and the corresponding values on the printed tube. The coordinate system has the origin in the centroid of the bottom cross-section. The  $z$  axis is coincident with the longitudinal axis of the digital model. For the digital model, the geometry of the tubular element is described by three scalar parameters only: the tube length ( $L_n$ ), the outer diameter ( $D_n$ ) and the thickness ( $s_n$ ).

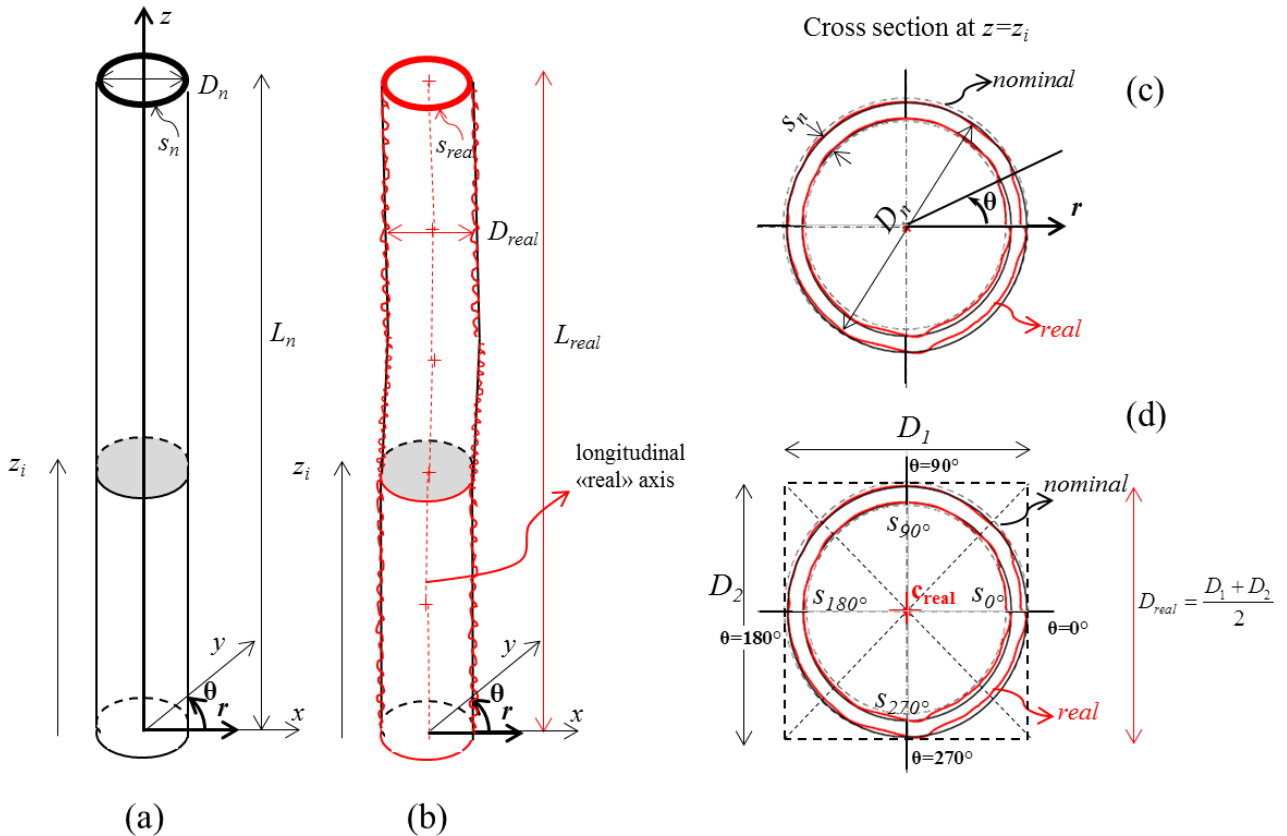


Figure 4: Digital model (a) and corresponding printed element (b) of a WAAM tube; polar coordinates along the section (c) and evaluation of outer diameter and thickness from printed outcome (d).

1 However, due to the intrinsic imperfections derived by the specific printing process, the outcome is  
2 a tubular element with a non-perfectly hollow circular cross-section which varies over the height and  
3 a non-straight longitudinal axis. In particular, both the real outer diameter  $D_{real}$  and the thickness  $s_{real}$   
4 vary with  $\theta$  and  $z$ , e.g.  $D_{real} = D_{real}(\theta, z)$  and  $s_{real} = s_{real}(\theta, z)$ . To reduce the variability of  $D_{real}$  to only  
5  $z$  variable, at each height  $z_i$  an equivalent outer diameter can be considered by constructing a  
6 circumscribed rectangle with the four sides that are tangent to the outer surfaces at  $\theta=0^\circ, 90^\circ, 180^\circ,$   
7  $270^\circ$ . The sides of the rectangle have lengths equal to  $D_1$  and  $D_2$ , so that  $D_{real} = \frac{D_1 + D_2}{2}$  is a function  
8 of  $z$  only, e.g.  $D_{real} = D_{real}(z)$ . Similarly, at each height  $z_i$  the thickness can be computed as the average  
9 over four measures taken at  $\theta = 0^\circ, 90^\circ, 180^\circ$  and  $270^\circ$ , so that  $s_{real} = \frac{s_{0^\circ} + s_{90^\circ} + s_{180^\circ} + s_{270^\circ}}{4}$ . At each  
10 height  $z_i$  the centroid of the tube ( $c_{real}$ ) is obtained as the intersection of the two diagonals of the  
11 circumscribed rectangle.

12

### 13 **3.2 Mechanical parameters**

14 The Wire-and-Arc Additive Manufacturing process (WAAM) is based on the deposition of fused  
15 welding material from a wire feedstock. In particular, MX3D printing process adopts 308LSi stainless  
16 steel as wire feedstock, therefore resulting in 3D-printed welded stainless steel elements.

17 Traditionally formed stainless steel is characterized, in general, by a highly non-linear behavior, thus  
18 requiring specific parameters to identify the mechanical behavior.

19 In 1943 Ramberg and Osgood ([32] Ramberg and Osgood 1943) first introduced a simplified equation  
20 to match the first part of the experimental stress-strain ( $\sigma$ - $\varepsilon$ ) curve, until a 0.2% proof stress ( $\sigma_{0.2}$ )  
21 which conventionally is assumed as the yielding point of stainless steel:

$$22 \quad \varepsilon = \frac{\sigma}{E_0} + 0.002 \left( \frac{\sigma}{\sigma_{0.2}} \right)^n \quad (1)$$

23 where  $n = \frac{\ln(20)}{\ln(\sigma_{0.2} / \sigma_{0.01})}$  (evaluated according to [33]) ensures that the Ramberg-Osgood equation

24 matches exactly the test values between 0.01% proof stress ( $\sigma_{0.01}$ ) and 0.2% proof stress ( $\sigma_{0.2}$ ), while  
25  $E_0$  is the initial Young's modulus that can be evaluated according to Eurocode 3-14 [34]. The second  
26 part of the stress-strain curve for stainless steel elements is usually evaluated according to Rasmussen  
27 equation [33].

28

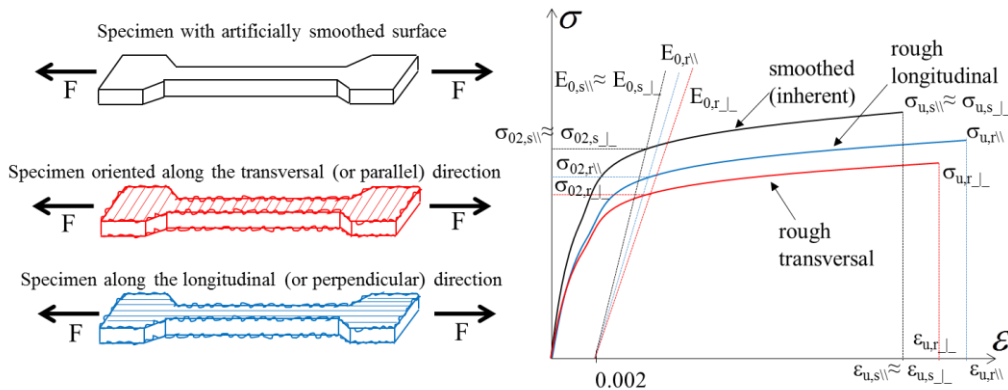
29

1 Traditionally manufactured 308LSi stainless steel grade attains mechanical properties in between  
 2 grade 304LSi and 316LSi, whose nominal values are given by various standards (see Table 1).

3 *Table 1: Summary of mechanical properties of grades 304LSi and 316LSi stainless steel for different standard provisions*  
 4 *[34, 35, 36].*

Grade		AS / NZS	SEI / ASCE	EC3 - 14
Austenitic 304 (EN 1.4301)	$E_0$ [GPa]	195	193.1	200
	$\sigma_{0.2}$ [MPa]	195 - 205	248.2 - 275.8	190 - 230
	$\sigma_u$ [MPa]	520	551.6 - 620.6	500 - 540
Austenitic 316 (EN 1.4401)	$E_0$ [GPa]	195	193.1	200
	$\sigma_{0.2}$ [MPa]	195 - 205	248.2 - 275.8	200 - 240
	$\sigma_u$ [MPa]	485	586.1 - 620.6	500 - 530

5  
 6 Since the manufacturing process may potentially induce an orthotropic behavior depending on the  
 7 orientation towards printing direction and the presence of surface roughness resulting from the  
 8 printing layers, the mechanical response will be investigated with reference to (1) samples with  
 9 artificially smoothed or as-manufactured surface, and (2) actions oriented along the printing direction  
 10 (also referred to as parallel direction) or perpendicular to it. Figure 5 shows a qualitative  
 11 representation of the different stress-strain response for three types of specimens subjected to tensile  
 12 action (F indicates the applied axial force). The following suffixes are used to indicate the mechanical  
 13 properties related to each type of samples: “s” standing for smoothed, “r” standing for rough, “\|”  
 14 standing for parallel, “\_” standing for perpendicular.



15  
 16 *Figure 5: Non-linear stress-strain curves of specimens with artificially smoothed surface and with rough surface oriented along the*  
 17 *printing (e.g. parallel) direction and along the orthogonal direction.*

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For the purpose of the present work, the following mechanical parameters will be investigated from the experimental results: 0.2% proof stress ( $\sigma_{0.2}$ ), ultimate stress ( $\sigma_u$ ), Young's modulus ( $E_0$ ), ultimate strain ( $\epsilon_u$ ).

**4. Design considerations for WAAM elements**

As introduced in the previous sections, structural elements manufactured with current WAAM processes are characterized by peculiar geometrical irregularities and specific material mechanical properties that have to be properly taken into account in both analysis and design process. Generally speaking, different approaches could be envisaged ranging from those based on advanced analysis and simulations able to model the real geometrical irregularities as well as material non-linearity, to simplified ones mainly developed for professional designers and compatible with current conventional analysis and design formats.

Theoretically speaking, the use of advanced simulation tools would allow to model the geometrical imperfections of each single manufactured piece and to consider the actual orthotropic non-linear stress-strain material behavior such that all the potential modes of failure could be explicitly included in the model. Such complex and detailed Finite Element models are also known as “digital twins” and would allow to even simulate loading tests, construction sequences, fatigue-related issues and other complex non-linear phenomena. These advanced analysis models may also be used along with structural monitoring systems for real-time control of the structural response. Pioneering research in this direction is currently under development by a research team led by Alan Turing Institute and MX3D in collaboration with Imperial College of London and Autodesk [37].

The full development of a reliable digital twin requires detailed knowledge of the peculiar geometrical imperfections requiring use of random field approaches and uncertainties quantification techniques ([38] Bae et al. 2004), as well as ad-hoc material models able to account for the specific features of WAAM. In order to properly manage such advanced simulation tools, structural engineers need to become more computationally literate and acquire high level computational skills ([14] Buchanan and Gardner 2019).

As an alternative to the advanced modelling of above, a simplified approach is here proposed. The idea lies upon decoupling the two main sources of variabilities: geometrical data and mechanical material properties.

As far as the geometrical characterization of each structural member is concerned, two types of results are derived: (1) general information on thickness and cross-sectional area then used for the

1 interpretation of the mechanical tests; (2) detailed information on surface roughness, lack of  
2 straightness, out-of-roundness, which could be used either for first engineering evaluations (within  
3 the scope of the present work) or for future works based on advanced modelling approaches.

4 As far as the mechanical material properties are concerned, the mechanical tests (developed with  
5 reference to the two main directions, i.e. parallel and perpendicular to the printing layer) are  
6 interpreted assuming that the cross-sectional area is constant along the member length so that the  
7 axial stresses can be computed according to simple beam theory:

$$8 \quad \sigma_{eff} = \frac{F}{A_{eff}} \quad (2)$$

9 Where:

- 10 •  $A_{eff}$  is an effective cross-sectional area of the structural member to be determined on the basis  
11 of the result type (1) of above.
- 12 •  $F$  is the tensile axial force (e.g. the force applied during a tensile test). Clearly, the stress  $\sigma_{eff}$   
13 can be interpreted as an effective stress which differs from the true material stress  
14 conventionally referred to as  $\sigma$ .

15 This result leads to condense all uncertainties (geometrical and mechanical) in the mechanical  
16 parameters only (see Section 6 for details). The simplified approach allows to treat the cross-sectional  
17 area as a deterministic value, while all the uncertainties are globally collected in the effective axial  
18 stress. In this way, the conventional format commonly adopted for the analysis and design of  
19 traditionally manufactured steel members, which considers the geometrical parameters as  
20 deterministic values and the material strength parameters as random variables, is maintained. Thus,  
21 the experimental mechanical parameters become dependent on both the specific geometrical and  
22 mechanical features related to the manufacturing process and not only on the material itself.

23 Clearly, the effective stresses  $\sigma_{eff}$  depend on  $A_{eff}$  and specific attention should be devoted to its  
24 choice. Different criteria can be adopted to chose  $A_{eff}$ , based on scientific, technical and practical  
25 considerations related to significance, accuracy and reliability of the chosen value. Some possibilities  
26 are (i) use of nominal values, (ii) use of a set of punctual values based on mechanical measurements,  
27 (iii) use of average values as obtained from volumetric measurements.

28 The use of nominal values as given by the manufacturer does not require any measurement, but, on  
29 the other hand, the absence of measurements may lead to lack of significance and poor reliability  
30 especially for non-standardized processes (like WAAM).

1 The use of few manual measurements (such as caliper measurements) is straightforward since such  
2 measurements can be easily executed even at the production site, but this type of measurements could  
3 be easily biased (see Sections 6.2 and 6.3).

4 The use of average values as obtained from volumetric measurements (used in the following proposed  
5 work to obtain  $A_{eff}$ , see details in Sections 6.2 and 6.3) has the advantage of providing an integral  
6 value based on equal weight criterion. It can also be adopted for a fast quality control during the  
7 production phase.

8

## 9 **5. The experimental campaign: methodological approach and work** 10 **program**

11 The work presents the first results of a wide experimental campaign carried out starting from January  
12 2017 (and still on-going) at the Topography and Structural Engineering Labs from University of  
13 Bologna. The aim is to assess the geometrical and mechanical properties of WAAM real-scaled  
14 specimens realized by MX3D using the process specifications mentioned in Section 2.

15

### 16 **5.1 The methodological approach: phases, measurements and mechanical** 17 **tests**

18 The phases of the experimental campaign have been defined in order to address the two most relevant  
19 issues from a structural design point of view, namely the intrinsic geometrical irregularities and the  
20 specific mechanical properties of WAAM elements.

21 The first phase is devoted to characterize the geometrical imperfections due to the adopted printing  
22 process, i.e. cross-section irregularities, surface roughness and lack of straightness. Since the  
23 evaluation of imperfections due to the manufacturing process may differ depending on the specific  
24 geometry to be printed, the specimens considered are both “dog-bone” shaped elements cut from  
25 rectangular plates and hollow circular tubes (stub columns and long tubes). Different types of  
26 measurements have been compared including: (i) punctual manual measurements with digital caliper  
27 with specific attention on the thickness for the “dog-bone” specimens and both thickness and outer  
28 diameter for the tubular specimens; (ii) volume-based measurements (from the Archimedes’  
29 principle) to obtain effective values of thickness for both types of specimens; (iii) 3D scan for a  
30 detailed evaluation of cross-section irregularities, surface roughness and lack of straightness of a  
31 single tubular element.

1 The second phase of the experimental campaign is devoted to the mechanical characterization of the  
 2 printed outcomes, making use of fundamental information gathered from the previous phase. For this  
 3 purpose tensile tests have been carried out on the “dog-bone” specimens cut from plates, compressive  
 4 tests have been carried out on stub column specimens, and buckling tests have been performed on  
 5 tubular elements. Photos of the tested specimens are shown in Figure 6.  
 6



7  
 8 *Figure 6: Specimens produced by MX3D: (a) rectangular plate; (b) “dog-bone” shaped coupons; (c) stub columns. (d) hollow*  
 9 *circular tubes.*

## 10 **5.2 The work program**

11 The entire experimental work program is summarized in Table 2, providing an overview of the  
 12 geometrical measurements and mechanical tests conducted on each sample. More in detail, the  
 13 columns of Table 2 give the following information:

- 14 • Type: CS indicates “dog-bone” shaped specimens with smoothed surface; CR indicates “dog-  
 15 bone” shaped specimens with rough surface; ST indicates short tubular specimens with rough  
 16 surface (stub columns); LT indicates long tubular specimens with rough surface.
- 17 • Nominal dimensions: thickness for “dog-bone” specimens type; diameter, thickness and  
 18 length for tubes type.
- 19 • Geometrical characterization: “Hand” indicates measurements with meter stick and caliper,  
 20 “Volume” indicates volume-based measurements using the Archimedes’ principle, “3D-  
 21 scan” indicates measurements obtained through 3D scanner.
- 22 • Mechanical characterization: “Tensile” refers to tensile tests on “dog-bone” specimens,  
 23 “Compression” refers to compression tests on stub columns, “Buckling” refers to the  
 24 buckling tests on long tubes.

- Series (number of specimens): the series identifies a group of identical specimens. They are distinguished using Roman numbers. Within each series, the single specimens are distinguished with Arabic numbers.
- Identification code ID: the ID uniquely identifies each single specimen. The code is composed of the following symbols and numbers: T/C/B indicates the type of experimental tests (T stands for Tensile, C for Compression, B for Buckling); the Roman numeral refers to the Series (from I to VI), the Arabic number identifies the specific specimen of that Series. “x” and “y” apply to “dog-bone” specimens only and indicate their orientation (“x” refers to the specimens cut along longitudinal direction, i.e. along the printing layer, “y” refers to those cut along the transversal direction, i.e. perpendicular to it).
- Presented results: reports the type of information and results provided for each series in the paper. “G” indicates that results of the geometrical characterization are reported for that specific series. “M” indicates that the results of the mechanical characterization are reported for that specific series. “N” indicates that no results are reported for that specific series.

Table 2: Summary of specimens characteristics and tests.

Type	Nominal dimensions	Geometrical characterization	Mechanical characterization	Series (number of specimens)	Specimen ID	Presented Results
CS	$t_n=4$ mm	Hand	Tensile	I (2)**	TI-1-y and TI-2-x	G+M
CR	$t_n=4$ mm	Hand	Tensile	II (4)**	From TII-1-x to TII-4-x	G+M
CR	$t_n=4$ mm	Hand	Tensile	II (4) **	From TII-1-y to TII-4-y	G+M
ST	$D_n=55$ mm, $s_n=5$ mm, $L_n=150$ mm	Hand Volume	Compression	III (9)	From CIII-1 to CIII-9	G+M
LT	$D_n=55$ mm, $s_n=5$ mm, $L_n=850$ mm	Hand Volume 3D-Scan *	Buckling	IV (3)	From-BIV-1 to BIV-3	G*
LT	$D_n=55$ mm, $s_n=5$ mm, $L_n=1050$ mm	Hand Volume	Buckling	IV (3)	From-BIV-4 to BIV-6	N
LT	$D_n=55$ mm, $s_n=5$ mm, $L_n=1200$ mm	Hand Volume	Buckling	IV (3)	From-BIV-7 to BIV-9	N



CS	$t_n=4$ mm	Hand Volume	Tensile	V (10) **	From TV-1-x to TV-10-x	N
CS	$t_n=4$ mm	Hand Volume	Tensile	V (10) **	From TV-1-y to TV-10-y	N
CR	$t_n=4$ mm	Hand Volume	Tensile	VI (20) **	From TVI-1-x to TVI-20-x	G
CR	$t_n=4$ mm	Hand Volume	Tensile	VI (20) **	From TVI-1-y to TVI-20-y	G

1 \* only specimen B-IV-3

2 \*\* “dog-bone” specimens of Series I, II, V, VI belong to the same batch (for a total of 10 plates).

3

4 More in detail, the geometrical characterization (first phase) encompassed the following steps:

- 5 a. Manual measurements on the first two Series (I and II) of “dog-bone” specimens cut from the  
6 rectangular plates;
- 7 b. Manual measurements and volume-based measurements on stub column specimens (Series  
8 III) and on the further series of “dog-bone” specimens (Series V and VI);
- 9 c. 3D scan of one tubular element (Series IV).

10 The analysis of part of the data obtained from the geometrical measurements is presented in Section  
11 6.

12 More in detail, the mechanical characterization (second phase) encompassed the following steps:

- 13 • Tensile tests on 2 “smoothed” (e.g. two specimens whose surfaces have been artificially  
14 polished) “dog-bone” specimens (Series I) and on 8 “rough” (e.g. with as-manufactured  
15 external surfaces) “dog-bone” specimens (Series II) to derive the first average mechanical  
16 properties of WAAM metal material and first indications of possible anisotropy related to  
17 the printing direction;
- 18 • Compression tests on 9 stub column coupons (Series III);
- 19 • Buckling tests on 9 tubular elements (Series IV) with different values of global slenderness;
- 20 • Tensile tests on 20 “smoothed” (Series V) and 40 “rough” “dog-bone” specimens (Series  
21 VI) using Digital Imaging Correlation (DIC) for a more reliable assessment of the  
22 mechanical properties and variability including the anisotropy due to the printing direction.

1 As far as the mechanical characterization is concerned, for the sake of conciseness, only selected  
2 results will be presented in this paper: (i) the results from the tensile tests on Series I and II are  
3 presented and discussed in Section 7; (ii) the results from the compression tests on Series III are  
4 presented and discussed in Section 8.

5 Detailed interpretation and discussion of buckling tests (Series IV) and of the further series of tensile  
6 tests (Series V and VI) will be presented in further separate works. Indeed, the aim of the present  
7 work is to provide first basic information related to geometrical properties and irregularities resulting  
8 from the manufacturing process (such as effective cross-sectional area, surface roughness, lack of  
9 straightness, out-of-roundness) and key mechanical properties (Young's modulus, 0.2% proof stress  
10 and ultimate stress, both under tension and compression). This first information could then be useful  
11 to develop guidelines and recommendations for structural design purposes.

12

## 13 **6. Geometrical characterization**

14 As introduced in Section 4, the geometrical characterization provided two types of results:

- 15 1. General information on thickness and cross-sectional areas based on manual and volume  
16 measurement (Sections 6.2 and 6.3), here used to obtain the values of effective stresses from  
17 the mechanical tests.
- 18 2. Detailed information on surface roughness, lack-of-straightness and out-of-roundness based  
19 on 3D scan (Section 6.4).

20 As far as results type 1 is concerned, one of the main purpose is the evaluation of the discrepancies  
21 between the *nominal* (associated to the digital model), the *real* (as manufactured) and the *effective*  
22 values (used to compute the effective stresses) of cross-sectional area. Figure 7 provides a qualitative  
23 comparison between nominal, real and effective cross-sectional area for a tubular element  
24 manufactured through WAAM.

25 Hereafter in this work the values of the effective cross-sectional areas are those derived from volume  
26 measurements. For each specimen, the effective cross-sectional area can be expressed in terms of the  
27 nominal value by means of the correction factor  $\varphi$  ( $A_{eff} = \varphi \cdot A_n$ ). In principle,  $\varphi$  could be treated as  
28 a random variable whose main descriptors (mean and coefficient of variation) can be determined from  
29 the results of geometrical measurements.

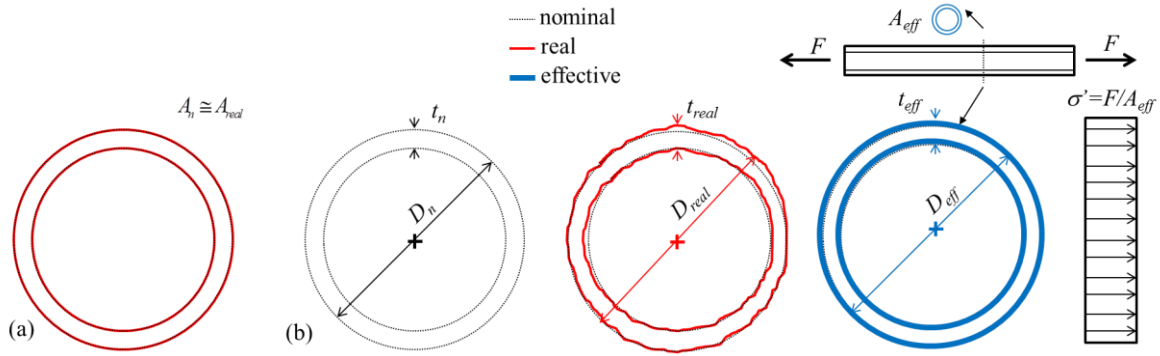
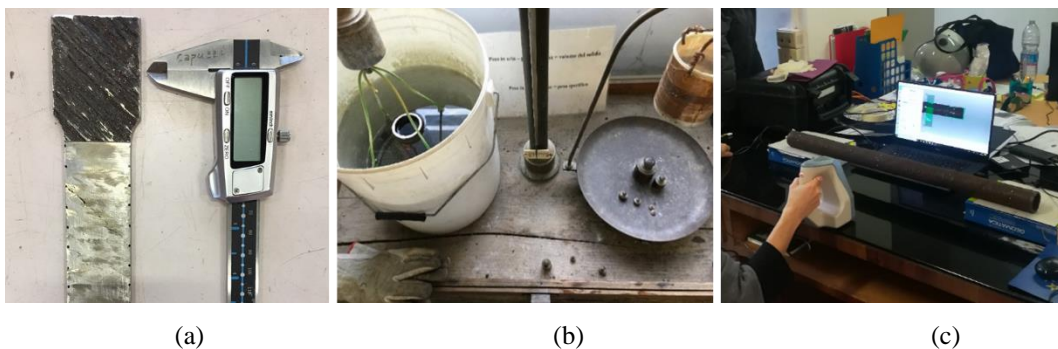


Figure 7: Tubular element: (a) manufactured with conventional techniques, (b) manufactured through WAAM (the difference between nominal, real and effective cross section is highlighted).

## 6.1. Measurements set-up

In order to assess the main geometrical parameters of Wire-and-Arc Additive Manufactured elements, the following measuring systems have been used:

- i. Digital caliper (with a nominal precision of 0.02 mm) for manual measures. Measurements include: (i) thickness of “dog-bone” shaped specimens cut from rectangular plates ( $t_c$ , where suffix “c” refers to caliper measures) at various locations, (ii) outer diameter of hollow circular coupons ( $D_c$ , where suffix “c” refers to caliper measures) by taking the average dimension of the circumscribed rectangle of the cross-section, as described in Section 3.1, (iii) thickness of hollow circular coupons ( $s_c$ , where suffix “c” refers to caliper measures) at different heights (Figure 8a);
- ii. Analogic hydraulic scale to take volume measurements from which the average thickness of both “dog-bone” shaped specimens ( $t_{av,v}$ , where suffix “v” refers to volume measures) and hollow circular coupons ( $s_{av,v}$ , where suffix “v” refers to volume measures) is derived (Figure 8b);
- iii. 3D scanning acquisition system for the evaluation of the cross-section irregularities, surface roughness and lack-of-straightness of one tubular element (Figure 8c).



1 *Figure 8: Measurement set-up for the geometrical characterization of WAAM elements: (a) caliper; (b) hydraulic scale; (c) Artec*  
2 *Spider 3D scanner.*  
3

## 4 **6.2. Thickness and cross-section of “dog-bone” specimens**

5 In order to characterize thickness ( $t_{real}$ ) and cross-sectional area ( $A_{real}$ ) of “dog-bone” shaped  
6 specimens, two types of measures have been performed:

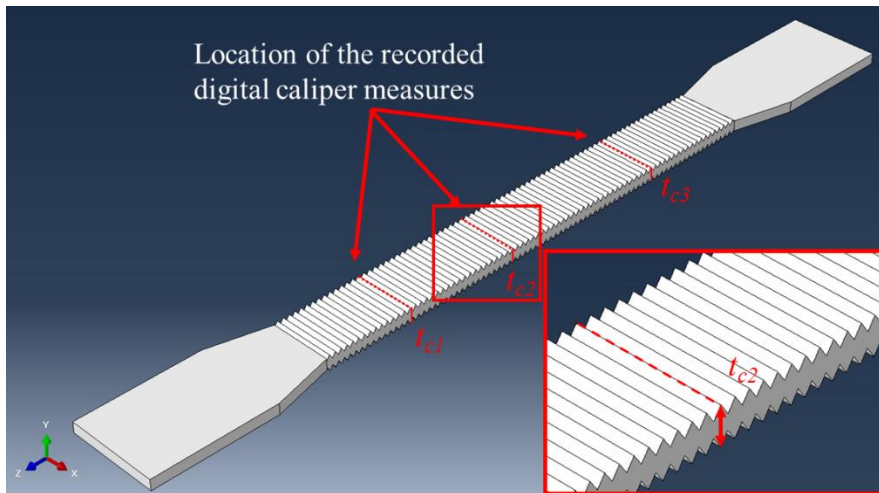
- 7 • Digital caliper measures of the main dimensions of the specimens and evaluation of thickness  
8 ( $t_c$ ) at 3 different sections of each specimen (see Figure 9) to identify the average thickness  
9 ( $t_{av,c}$ ). These measurements (see Table A1 in Appendix A) have been performed on all  
10 specimens of Series I, II (10 in total) and VI (40 in total).
- 11 • Volume measures from the Archimedes’ principle, taking the weight on air ( $m_{air}$ ) from a  
12 digital scale and the weight inside water ( $m_{H_2O}$ ) from an analogic hydraulic scale. Based on  
13 the two measures taken on the weight, the specimens’ volume has been derived from the  
14 calculated values of material density  $\gamma_v$  (see Table A2 in Appendix A) and density of water  
15 (at 25°C)  $\gamma_{H_2O}$  with the following formulation:  $V = (m_{air} - m_{H_2O}) / \gamma_{H_2O}$ . Then, the real average  
16 thickness ( $t_{av,v}$ ) of the specimen has been computed dividing the volume by the specimen  
17 surface area (computed from the measures of the external geometry taken with digital caliper).  
18 These measurements have been performed on Series VI specimens (40 in total).

19 For each specimen of Series VI, the ratio  $\chi$  between the average thickness obtained from caliper  
20 measurements ( $t_{av,c}$ ) and that from volume measurements ( $t_{av,v}$ ) has been computed.

21 Tables A1 and A2 in Appendix A report all the measured values for Series I, II and VI. For Series I,  
22 the mean value of thickness (over 2 specimens) as obtained from caliper measurements results equal  
23 to 2.92 mm, while for Series II the mean (over 8 specimens) results equal to 4.38 mm with a  
24 coefficient of variation of 0.04. For Series VI, the mean value of thickness (over 40 specimens) as  
25 obtained from caliper measurements results equal to 4.52 mm with a coefficient of variation equal to  
26 0.07. The mean value of thickness as obtained from volume measurements for specimens of Series  
27 VI results equal to 3.92 mm with a coefficient of variation equal to 0.10.

28 The measurements indicate that, on average, the thickness as obtained from caliper measurements  
29 ( $t_{av,c}$ ) is slightly larger (+13%) than the nominal value ( $t_n=4$  mm). On the contrary, the average value  
30 of thickness as obtained from volume measurements ( $t_{av,v}$ ) is slightly less (-2%) than the nominal  
31 value ( $t_n$ ). For Series VI, the average value of coefficient  $\chi$  results equal to  $\chi_{av}=1.16$ . The  
32 discrepancy between caliper measures and volume measures results from the nature of the surface  
33 roughness, having very sharp peaks and troughs as qualitatively shown in Figure 9. Indeed, the caliper

1 measures are representative of thickness values taken around the peaks, given the practical difficulty  
 2 of obtaining measurements close to the troughs. As a consequence, the resulting average values ( $t_{av,c}$ )  
 3 tend to be larger than those obtained through volume measurements, which can be considered as some  
 4 sort of integral values over the entire surface, thus accounting for both contributions of the peaks and  
 5 troughs. In addition, the difference between the average thickness from volume measurements and  
 6 the nominal thickness can be interpreted as a systematic error due to limited precision of the  
 7 manufacturing process. In light of this, the values of thickness derived from volume measurements  
 8 can be considered as effective values, since they correspond to those of an equivalent specimen with  
 9 constant (effective) thickness having equal volume. The ratio between the average values of thickness  
 10 from volume measurements  $t_{av,v}$  and the nominal thickness  $t_n$  represents the mean value of the  
 11 correction factor  $\varphi$  which results equal to  $\varphi_{av}=0.78$ , with a coefficient of variation of 0.10.  
 12 Regarding Series II specimens, for which no volume measurement was taken, the value of the  
 13 effective cross-sectional area (then used for the calculation of the stresses, see Section 7) has been  
 14 obtained dividing the average value obtained from caliper measurements ( $A_{av,c}$ ) by the factor  $\chi_{av}$   
 15  $=1.16$  (calibrated based on measurements on specimens from Series VI). The assumption is justified  
 16 by the common nature of the surface roughness of Series II and Series VI, as all plates were  
 17 manufactured with the same printing strategy and deposition method.



18  
 19 *Figure 9: Location of the digital caliper measurements and detail of one measurement.*  
 20

### 21 **6.3. Thickness, cross-section and outer diameter of stub columns coupons**

22 In order to characterize the geometry (cross-sectional area and thickness) of 9 stub columns coupons  
 23 (Series III), two types of measures have been performed, following the same approach described in  
 24 the previous Section:

- Digital caliper measures of outer diameters ( $D_c$ ) and thickness ( $s_c$ ) taken at 3 different heights  $z_i$  (e.g. 40 mm, 80 mm and 120 mm) to identify the average real outer diameter ( $D_{av,c}$ ) and cross-sectional area ( $A_{av,c}$ ).
- Volume measures from the Archimedes' principle, taking the weight on air ( $m_{air}$ ) from a digital scale and the weight inside water ( $m_{H2O}$ ) from an analogic hydraulic scale. Based on the same procedure as the one described in Section 6.3, from the measures of the specimens volume and length ( $L_c$ ), the average cross-sectional area  $A_{av,v}$  has been obtained. Then, based on the values of the outer diameter taken from digital caliper measures ( $D_c$ ) and assuming circular hollow cross-section, the average thickness ( $s_{av,v}$ ) of the specimen has been derived.

For each specimen, the ratio  $\chi$  between the average thickness as obtained from caliper measurements ( $s_{av,c}$ ) and from volume measurements ( $s_{av,v}$ ) has been computed.

Table A3 in Appendix A provides the results from the measures listed for each of the 9 specimens of Series III. The mean value (over the 9 specimens) of the average diameter  $D_{av,c}$  as obtained from caliper measurements results equal to 53.97 mm. The coefficient of variation results less than 0.01, indicating a quite reduced variability with respect to the other geometrical quantities here evaluated. The maximum discrepancy between the measured average diameter  $D_{av,c}$  and the nominal outer diameter ( $D_n = 55$  mm) is equal to 1.40 mm corresponding to a relative error of 2.50%. It should be noted that the tolerances in terms of outside dimensions for cold-formed welded circular hollow sections as given by various standard provisions ([39] EN 10219-2:2006, [40] EN 10162:2003, [41] EN 10051:2010) are  $\pm 1$  % with a minimum of  $\pm 0.3$  mm and a maximum of  $\pm 1.0$  mm.

The mean value of thickness (over the 9 specimens) as obtained from caliper measurements  $s_{av,c}$  results equal to 5.25 mm, with a coefficient of variation of 0.04. The mean value of thickness (over the 9 specimens) as obtained from volume measurements  $s_{av,v}$  results equal to 3.81 mm, with a coefficient of variation of 0.03. For each single specimen, the thickness from volume measurement ( $s_{av,v}$ ) is obtained from the measured volume and the average outer diameter  $D_{av,c}$ . Then, the average outer diameter ( $D_{av,c}$ ) is used together with the thickness from volume measurement to compute the volume-based area ( $A_{av,v}$ ), assumed as effective area. Overall, the thickness values as obtained from digital caliper measures ( $s_{av,c}$ ) are, on average, slightly (+5%) larger than the nominal value ( $s_n=5$ mm). On the contrary, the thickness values (e.g. the effective thickness) as obtained from volume measurements ( $s_{av,v}$ ) are, on average, smaller (-23.70%) than the nominal value. Moreover, the mean value of the correction factor of the cross-sectional area ( $\varphi$ ) results equal to  $\varphi_{av}=0.85$ . The coefficient of variation is equal to 0.02. The larger discrepancy between the nominal and effective thickness for the stub columns coupons (-23.70%) can be interpreted as a sort of systematic error

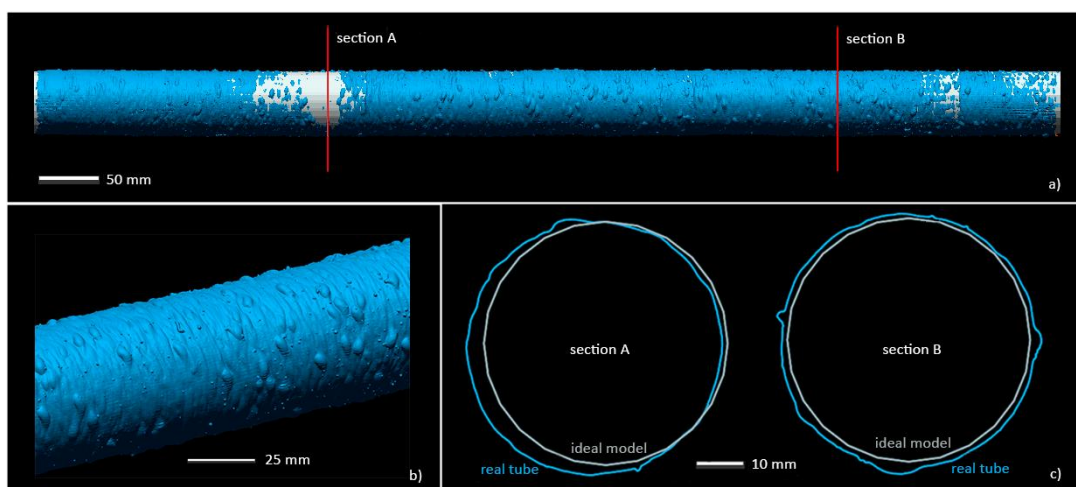
1 rather than a consequence of random variability, since the coefficient of variation of volume  
2 measurements for stub columns coupons results smaller than the one of “dog-bone” specimens  
3 (COV=0.03 for stub columns coupons against COV=0.10 for “dog-bone” specimens of Series VI).  
4 The stress values computed from the compressive test results (described in Section 8) are based on  
5 the volume measurements ( $A_{av,v}$ ), considered as effective.  
6

#### 7 **6.4. 3D scan of one tube (BIV-3)**

8 One of the WAAM tubular elements (e.g. specimen BIV-3) then tested for buckling has been first  
9 scanned by means of high resolution 3D scanning. The equipment consists of an optical non-contact  
10 measuring system based on the blue light, therefore on the principles of topographic triangulation.  
11 The projected fringe pattern of blue light on the object surface evidences distortions due to the  
12 specimen’s surface shape. For the present work, two cameras have been adopted to capture the  
13 distortion and calculate the corresponding 3D coordinate measurements, acquiring full-field scans of  
14 a volumetric area and collecting millions of points per scan.

15 The structured-light projection Artec Spider scanner ([42] <https://www.artec3d.com>) adopted for the  
16 present work has an acquisition speed of 1 mln points/sec for metrology applications. The 3D  
17 resolution obtained is of 100 points/mm<sup>2</sup>, with a 3D point accuracy of 0.05 mm for a medium field  
18 size of 90x70 mm and a working distance of 0.25 m.

19 The obtained 3D model of the tubular specimen consists of around 40 million of triangular elements,  
20 with a medium size of about 0.10 mm. Figure 10 shows a view of the entire mesh of the real tube  
21 (blue model), as compared with the digital model (grey model), as well as a zoom of a portion to  
22 better appreciate the surface roughness and two cross-sections referred to as sections A and B.  
23

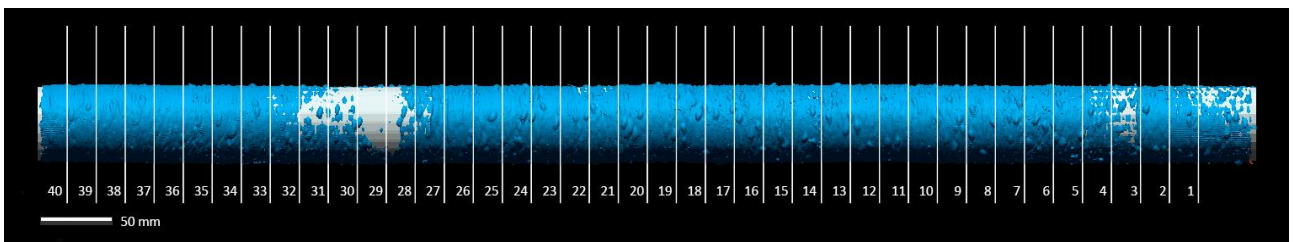


24

1 *Figure 10: a) 3D model of the real printed tube (blue) obtained by 3D scanning, compared with the digital model (grey);*  
2 *b) detail of the model; c) sections of the two models.*

3  
4 The data obtained from the 3D scan of one tubular specimen (e.g. BIV-3) have been processed to  
5 obtain a first characterization of the main external surface irregularities (such as variation of the outer  
6 diameter along the length and variation of the surface roughness both along the length and with  
7 respect to angle  $\theta$ ) and of the global geometrical imperfections (such as lack of straightness, out-of-  
8 roundness) relevant from a structural design perspective. For this aim, a number of 40 cross-sections  
9 spaced at 20 mm along the longitudinal axis extracted from the complete 3D model (Figure 11) has  
10 been analyzed. Clearly, for an accurate study of the surface features a more refined discretization  
11 would be required (providing data at cross-sections spacing of the order of 0.5 mm so that both peaks  
12 and troughs would be properly captured). Nevertheless, such detailed geometrical analysis is out of  
13 the scope of the present paper.

14 Each of the 40 cross-sections has been analyzed according to the schematization of Figure 4 with the  
15 purpose of obtaining the values of  $D_s$  (suffix “s” stands for the measures taken from 3D scanner  
16 acquisition) and the coordinates of the centroid  $C_{real}=(C_{real,x}, C_{real,y})$ . Note that at each cross-section the  
17 outer diameter  $D_s$  has been taken from the dimensions of the circumscribed rectangle, as described in  
18 Section 3.1. Since only the external surface has been scanned, the coordinates of the centroid have  
19 been computed as the center of the circumscribed rectangle, thus implicitly assuming a constant wall  
20 thickness for each cross section. Details of the measures taken at each cross-section are listed in Table  
21 A4 of Appendix A.



23 *Figure 11: Model of real tubular element from 3D scanning acquisition.*

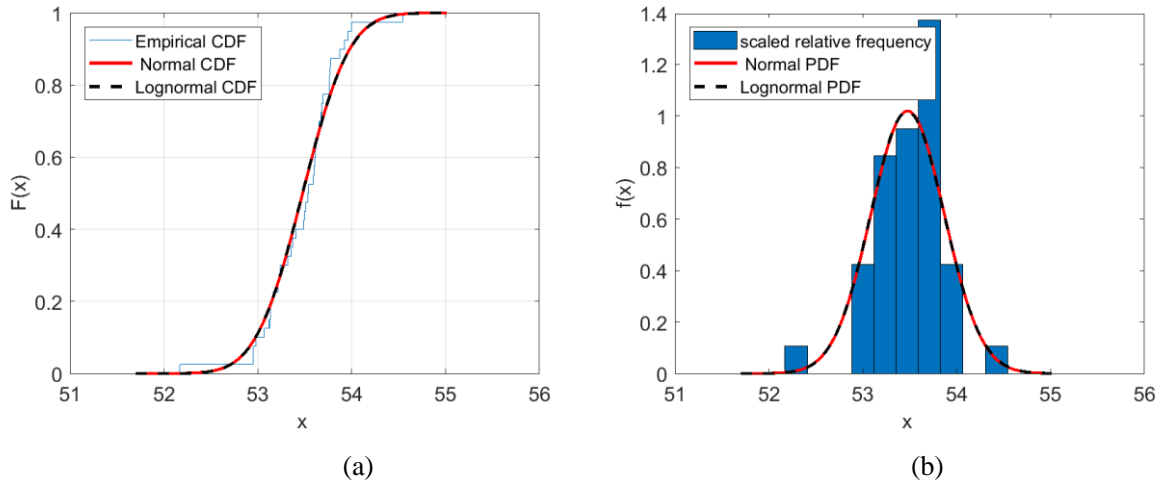
#### 24 25 26 6.4.1. Outer diameter

27 From the evaluation of the 40 cross-sections taken on the 3D scan, the resulting mean and standard  
28 deviation of  $D_s$  along the tubular element are equal to  $\mu_{D_s}=53.48$  mm and  $\sigma_{D_s}=0.39$  (corresponding  
29 to a COV=0.01). On average, the relative error in the real outer diameter with respect to the digital  
30 model, e.g.  $(D_n-\mu_{D_s})/D_n$ , is equal to 2.76%. Figure 12 shows that the experimental CDF of the real



1 outer diameter can be well fitted by a Gaussian distribution (for the sake of comparison also the Log-  
 2 normal distribution which fits the data has been plotted in the graph).

3



4

5

6 *Figure 12: Distribution of outer diameter values: (a) cumulative distribution functions (CDFs); (b) probability density functions*  
 7 *(PDFs).*

8

9

#### 6.4.2. Surface roughness

10 The data of the 3D scan have also been used to analyze the surface roughness of the tubular element.  
 11 In particular, for each of the 40 selected cross-sections, 4 measures of the outer radius  $r$  have been  
 12 taken, each one at a different angle  $\theta$  (e.g.  $\theta=0^\circ$ ,  $\theta=90^\circ$ ,  $\theta=180^\circ$ ,  $\theta=270^\circ$ ) as shown in Figure 4.  
 13 Figure 13 displays all the measured values of the outer radius along the longitudinal axis,  
 14 distinguishing each measurement angle with a different color. The thin continuous black line indicates  
 15 the constant value of the radius in the digital model, equal to  $D_n / 2 = 27.5$  mm, while the thick dotted  
 16 black line corresponds to the average value of the outer radius at the 40 sections considered, equal to  
 17  $D_{av,s} / 2 = 26.74$  mm. The discrepancies between the real outcome of the printing process and the digital  
 18 model vary along the  $z$ -axis and with respect to the angle of rotation  $\theta$ , thus indicating a non-uniform  
 19 roughness both along the length of the element and along the cross-section at each cutting plane.  
 20 However, it should be noted that the discrepancy between the average value obtained from 3D  
 21 scanning data ( $D_{av,s} / 2$ ) and the digital model ( $D_n / 2$ ) is of around 0.75 mm (corresponding to a  
 22 relative error of 2.70%).

23

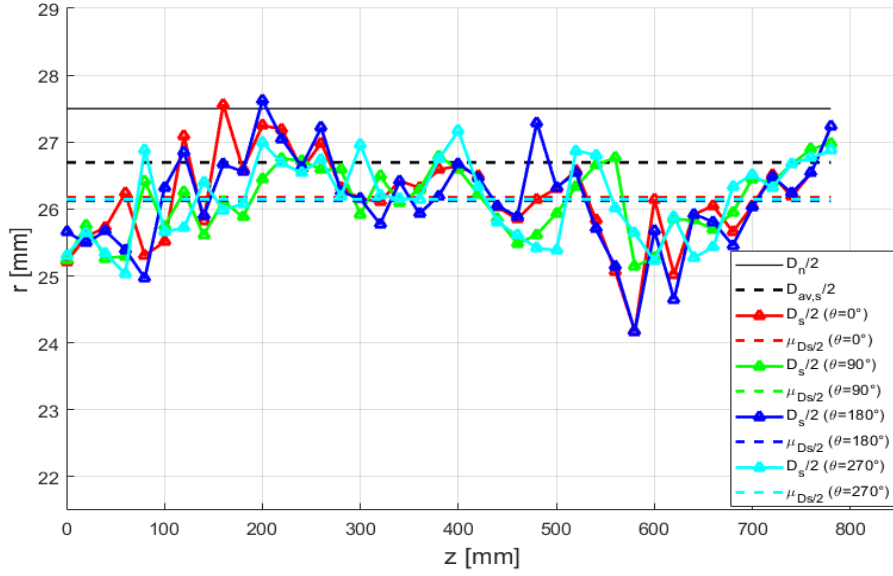


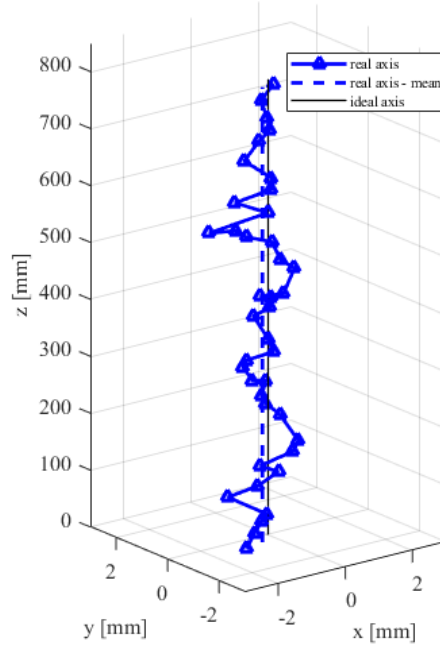
Figure 13: Surface roughness along longitudinal axis over 4 points of the external surface: at  $\theta=0^\circ$ ,  $\theta=90^\circ$ ,  $\theta=180^\circ$ ,  $\theta=270^\circ$ .

#### 6.4.3. Lack of straightness

The data obtained from the 3D scanning are also used to evaluate the lack of straightness of the longitudinal axis (along  $z$  direction), considering the same 40 cross-sections of Figure 11. For each cross-section the lack of straightness is computed as the distance between the  $z$ -axis and the centroid of the circumscribed rectangle ( $C_{real}$ ). The distance corresponds to the coordinates of the centroid ( $C_{real,x}$ ,  $C_{real,y}$  using Cartesian  $x$ - $y$  coordinates or  $C_{real,r}$  using cylindrical coordinates). Figure 14 displays the piece-wise line that connects the centroids of the 40 cross-sections ( $C_{real}$ ), which can be defined as the “real longitudinal axis” of the printed tube (see Figure 4), while Table 3 provides numerical values of maximum, minimum and average discrepancies between the real longitudinal axis (thick blue line) and the  $z$ -axis (thin black line). The maximum (absolute value) of the lack of straightness is of  $L/300$ , while the average is of around  $L/1000$ .

It is important to note that steel beams/columns manufactured with conventional technologies (laminated profiles) generally have minimum geometrical imperfections in the order of  $L/1000$  ([40] EN 10162:2003). In addition, for cold-formed welded tubular members, the tolerance in straightness is of  $L/500$  according to EN 10219-2 [38]. It should be remarked that, in general, the lack of straightness depends upon two contributions: (1) global crookedness due to WAAM nature (layer-by-layer deposition); (2) possible local centroid deviation due to variation of wall thickness with  $\theta$  (Figure 4c). Nonetheless, at first approximation, the contribution due to local centroid deviation could be considered negligible with respect to the global crookedness. Therefore, the values computed are based on the assumption of constant wall thickness for each cross-section since the centroid

1 coordinates have been determined as the center of the circumscribed rectangles. For a more precise  
 2 evaluation of the local variation in the thickness, a complete 3D scanning of the inner part of the tube  
 3 would be required as well.



4  
5  
6  
7  
*Figure 14: Study of straightness of longitudinal axis.*

*Table 3: Summary of study of longitudinal axis.*

<b>Longitudinal axis</b>	<b><math>c_{real,x}</math> [mm]</b>	<b><math>c_{real,y}</math> [mm]</b>	<b><math> c_{real,r} </math> [mm]</b>	<b><math>c_{real,x}/L</math> [%]</b>	<b><math>c_{real,y}/L</math> [%]</b>	<b><math> c_{real,r} /L</math> [%]</b>
max	0.51	0.38	2.52	0.06	0.05	0.32
min	-2.39	-1.44	0.06	-0.30	-0.18	0.01
<b><math>\mu</math></b>	<b>-0.46</b>	<b>-0.38</b>	<b>0.81</b>	<b>-0.06</b>	<b>-0.05</b>	<b>0.10</b>
<b>COV</b>	<b>1.23</b>	<b>1.22</b>	<b>0.61</b>	<b>1.23</b>	<b>1.22</b>	<b>0.61</b>

8  
9 Figure 15 shows that the experimental CDF of the real longitudinal axis is quite asymmetric and  
 10 therefore can be well fitted by a Log-normal distribution (for the sake of comparison also the Gaussian  
 11 distribution fitting the experimental data is plotted in the graph). It can be noted that the Gaussian  
 12 distribution does not well fit the experimental data. This is due to a significant asymmetry and very  
 13 large coefficient of variation (around 0.60) of the experimental data.

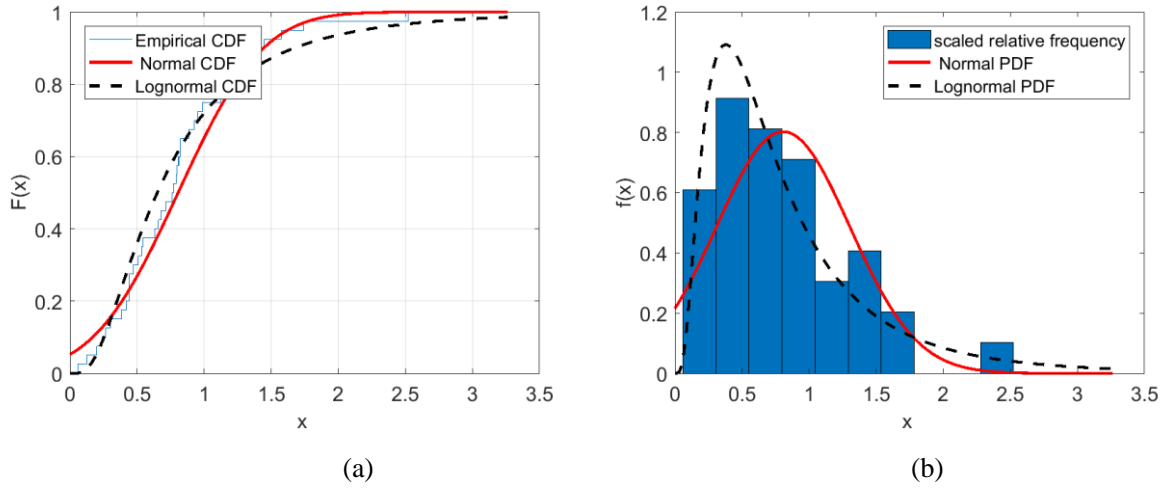


Figure 15: Distribution of longitudinal axis values: (a) cumulative distribution functions (CDFs); (b) probability density functions (PDFs).

#### 6.4.4. Out-of-roundness

The out-of-roundness ( $O$ ) values have been computed considering the 40 cross-sections of Figure 11. For each cross-section, out-of-roundness values are computed using  $D_1$  and  $D_2$  values (see Figure 4)

according to the following equation:  $O(\%) = \frac{|D_2 - D_1|}{D_n} \cdot 100$ . The mean value of  $O(\%)$  along the 40

cross-sections results equal to 1.31%, with a coefficient of variation of 0.70. The maximum value of  $O(\%)$  results equal to 3.20%. Figures 16 shows the experimental CDF and the relative frequency histogram of the out-of-roundness ( $O$ ). It can be noted that the data are quite uniformly distributed within the domain. For traditionally cold-formed welded circular hollow cross-sections the maximum out-of-roundness tolerance is equal to 2% according to EN 10219-2 [38]. ISO 19902 [43] provides maximum out-of-roundness tolerance of 3% for fixed steel offshore structures.

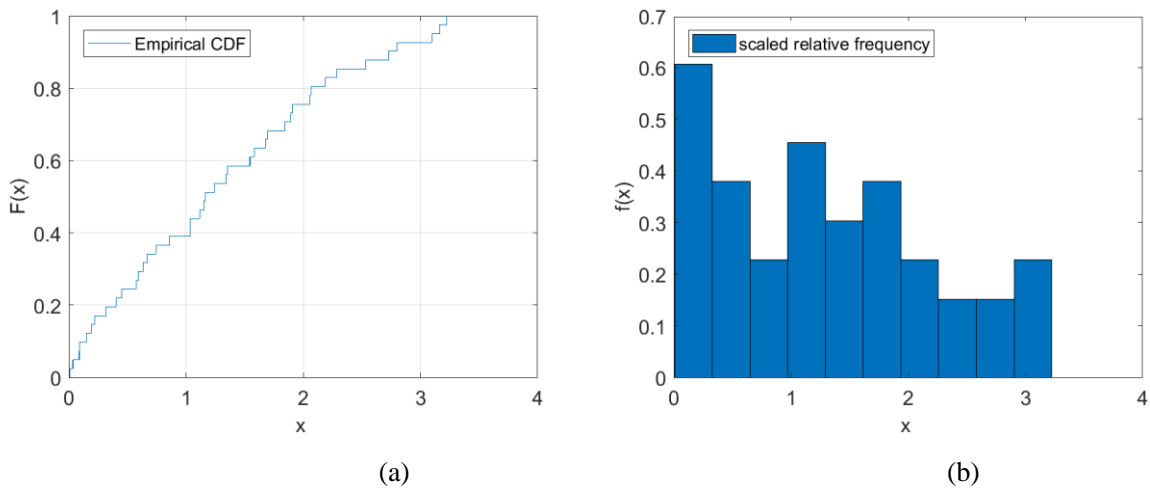


Figure 16: Distribution of out-of-roundness values: (a) cumulative distribution functions (CDFs); (b) relative frequency histogram.

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## **7. Tensile coupon tests (series I and II)**

The first experimental campaign aimed at assessing the mechanical behavior of the printed material under tensile loading has been carried out in 2017 over a number of 10 samples (Series I and II). The preliminary results of part of the experimental tests have been presented in a previous work done by the authors ([44] Laghi et al. 2018).

### **7.1. Specimens and set-up**

The tensile test specimens are cut along the two main directions of the Additive Manufactured stainless steel plates, as described in Section 3.1 (see Figure 3), following the provisions in terms of shape and dimensions for tensile test samples described in ISO 6892-1:2016 ([45]). The longitudinal direction, referred to as  $x$ , has been taken parallel to the printed layers, while the transversal direction, referred to as  $y$ , perpendicular to that. The response from the different orientation of the specimens would give first useful information regarding the anisotropy and build orientation influence of WAAM manufacturing technique.

Among the 10 specimens tested, the specimens from Series I (TI-1-y and TI-2-x) have been previously artificially polished with milling cutter in order to reduce the surface roughness and thus have a first valuation of the mechanical properties on the material itself, reducing the influence of geometrical surface irregularities.

The tests have been performed using a Universal Machine of 600 kN nominal tensile capacity by applying the tensile load at a rate of 1 MPa / sec. The longitudinal elongation of each specimen is measured using a Linear Variable Differential Transformer (LVDT). Figure 17 shows the adopted testing set-up.

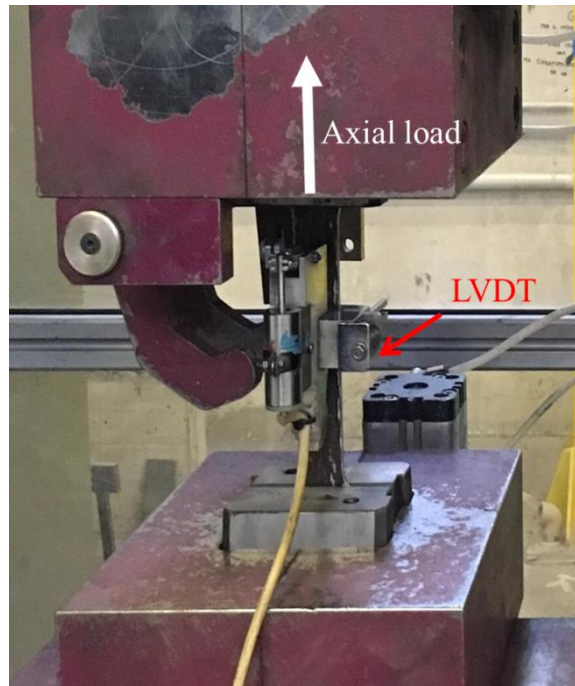


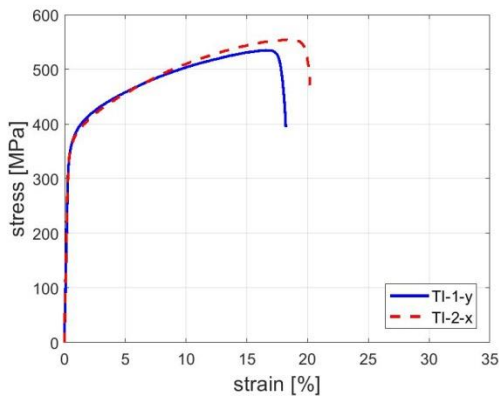
Figure 17: Experimental set-up of the tensile tests on “dog-bone” specimens.

## 7.2. Test results

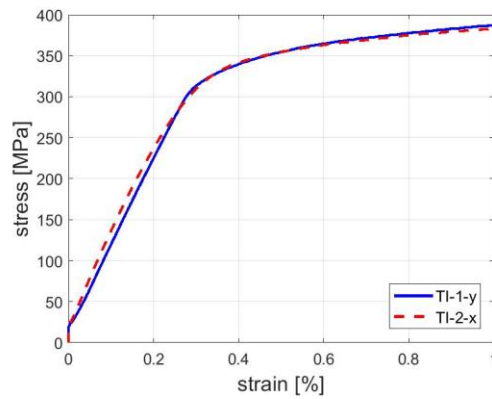
The stress-strain diagrams as obtained from all tensile tests of Series I and II are shown in the graphs of Figure 18, separating the two “smoothed” specimens of Series I (Fig. 18 a and b), from the “rough” specimens of Series II, oriented along the longitudinal direction ( $x$ ) (Fig. 18 c and d) and along the transversal direction ( $y$ ) (Fig. 18 e and f). As explained in Section 6.2, for the specimens belonging to Series II, the stresses are obtained dividing the recorded values of loads by the effective cross-sectional area assumed to be equal to  $A_{av,c} / \chi_{av}$ . Values of strains are taken from the measures recorded by the LVDT. Table 4 collects the numerical values of the main mechanical parameters.

Overall, the stress-strain material behavior presents a fundamental difference with respect to conventional structural stainless steel material in terms of Young’s modulus ( $E_0$ ), whose values result to be between 108 to 152 GPa, with an average of 115 GPa. This corresponds to 60% of the usual Young’s modulus of cold-formed stainless steel suggested by common standard provisions [34, 35, 36] (equal to 200 GPa, as shown in Table 1). The reduced Young’s modulus values observed for WAAM material might be due to several factors: (i) the intrinsic residual stresses given by the temperature gradient between added layers of welded material; (ii) the different microstructure of Wire-and-Arc Additive Manufactured metal material, as highlighted by recent research works ([46, 47]). However, further investigation on these factors is needed in order to give proper explanation of such behavior. On the other hand, the values of the other significant response parameters (0.2% proof

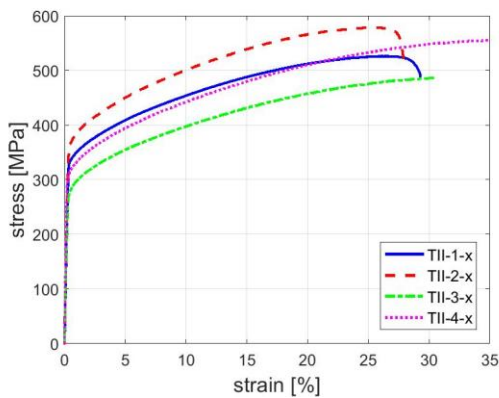
1 stress  $\sigma_{0.2}$  and ultimate stress  $\sigma_u$ ) are closer to those of traditionally manufactured stainless steel (see  
 2 Table 1). In particular, the values of 0.2% proof stress  $\sigma_{0.2}$  are larger (+50%) than those given by the  
 3 codes reported in Table 1, while the ultimate stresses  $\sigma_u$  are within the ranges.  
 4 As far as the behavior along the two directions is concerned, some discrepancies can be noted. First  
 5 of all, from the graphs providing the stress-strain responses at low strains (<1%, e.g. Fig. 18 b, d, f)  
 6 the response of the specimens cut along the transversal direction (y) appears smoother, as more  
 7 rounded stress-strain response can be appreciated already at low values of stresses. Moreover,  
 8 specimens cut along the transversal direction (y) result in lower mechanical properties when  
 9 compared to those cut along the longitudinal direction (x). In more details, the reductions are: 13%  
 10 for 0.2% proof stresses  $\sigma_{0.2}$ , 9% for ultimate stresses  $\sigma_u$ , 19% for Young's modulus  $E_0$  and 32% for  
 11 ultimate strain  $\epsilon_u$ .



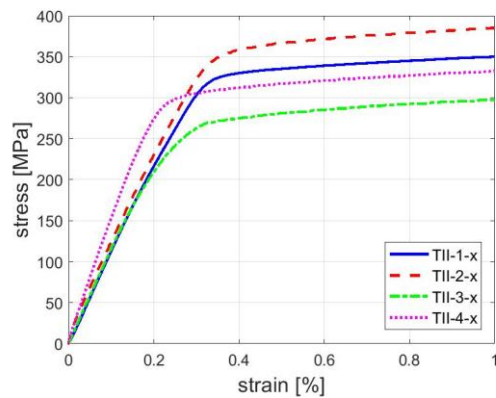
(a)



(b)



(c)



(d)

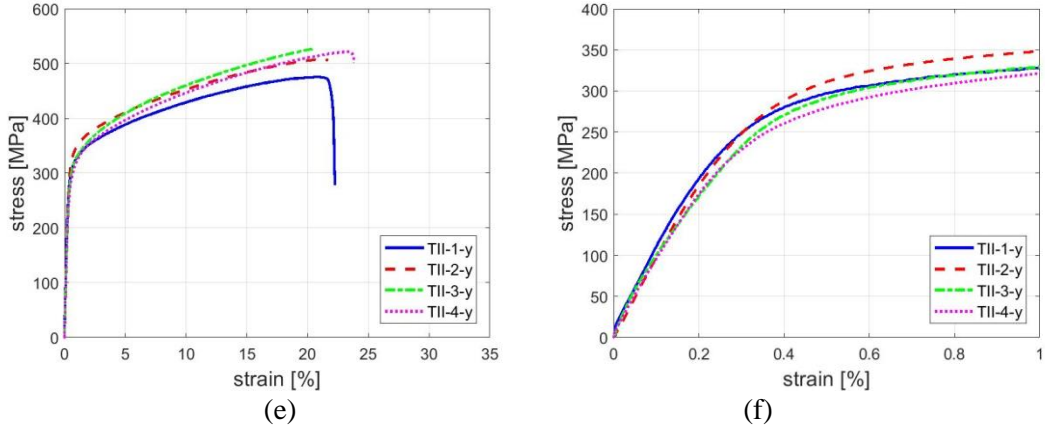


Figure 18: Stress-strain curve from tensile tests data. Series I: (a) full response; (b) zoom for strain < 1%. Series II: (c) full response (specimens oriented along x direction); (d) zoom for strain < 1% (specimens oriented along x direction); (e) full response (specimens oriented along y direction); (f) zoom for strain < 1% (specimens oriented along y direction)

Table 4: Results from tensile tests (Series I and II).

Specimen	$\sigma_{0.2}$ [MPa]	$\sigma_u$ [MPa]	$\epsilon_u$ [%]	$E_0$ [GPa]
TI-1-y	355	535	18	108
TI-2-x	353	554	20	113
TII-1-x	364	558	29	119
TII-2-x	384	613	28	118
TII-3-x	302	517	33	121
TII-4-x	338	588	40	152
$\mu$	<b>347</b>	<b>569</b>	<b>33</b>	<b>128</b>
COV	<b>0.08</b>	<b>0.06</b>	<b>0.13</b>	<b>0.10</b>
TII-1-y	315	505	22	106
TII-2-y	333	539	22	102
TII-3-y	267	485	20	101
TII-4-y	298	553	24	105
$\mu$	<b>303</b>	<b>521</b>	<b>22</b>	<b>104</b>
COV	<b>0.07</b>	<b>0.05</b>	<b>0.06</b>	<b>0.02</b>

## 8. Stub column tests

The second part of the experimental campaign presented in Section 5.2 has been devoted to assess the behavior in compression of Wire-and-Arc Additive Manufactured stainless steel material, with a total number of 9 stub columns tested under monotonic compressive load (Series III).



### 8.1. Specimens and set-up

The stub columns have a nominal length  $L_n$  of 150 mm, a nominal outer diameter  $D_n$  of 55 mm and a nominal thickness  $s_n$  of 5 mm from the digital model. The tests are conducted with the same Metrocom testing machine used for the tensile tests (Figure 19). The monotonic compressive load is applied at a constant speed of 1 MPa / sec. The longitudinal elongation of the specimen is measured using two Linear Variable Differential Transformers (LVDTs) positioned at two sides.



Figure 19: Experimental set-up of the compression tests on stub column specimens.

The main aim of the tests is to investigate the behavior up to yielding in order to compare the values of Young's modulus and 0.2% proof stress with those obtained from the tensile tests. Indeed, due to the initial out-of-straightness, the specimens can be subjected to combined compression and bending mainly influencing the post-elastic behavior and local buckling mode of failure. Figure 20 evidences the deformed configurations at the end of the test for three different specimens. It can be noted that the local buckling modes of failure are quite different from each other. However, detailed interpretation of the failure modes is out of the scope of the present work.

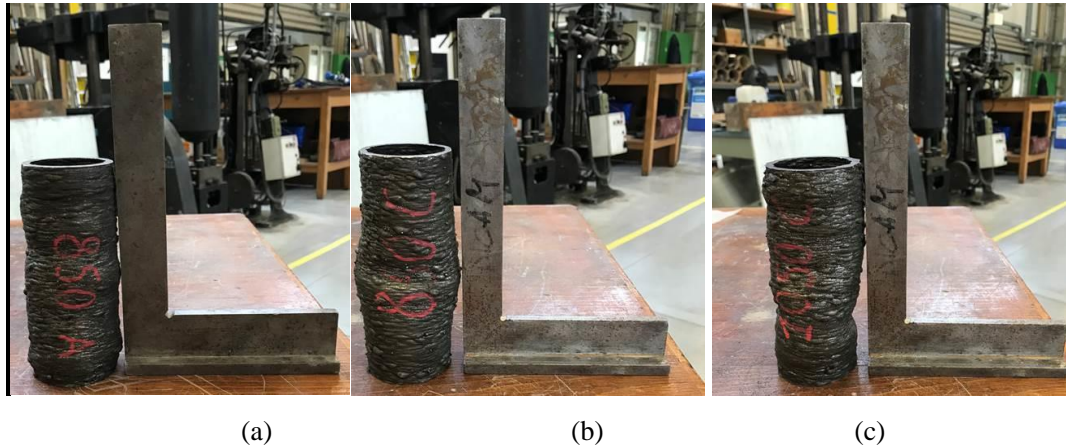


Figure 20: Deformed configuration of stub column coupons at the end of the test: (a) specimen CIII-1; (b) specimen CIII-3; (c) specimen CIII-6.

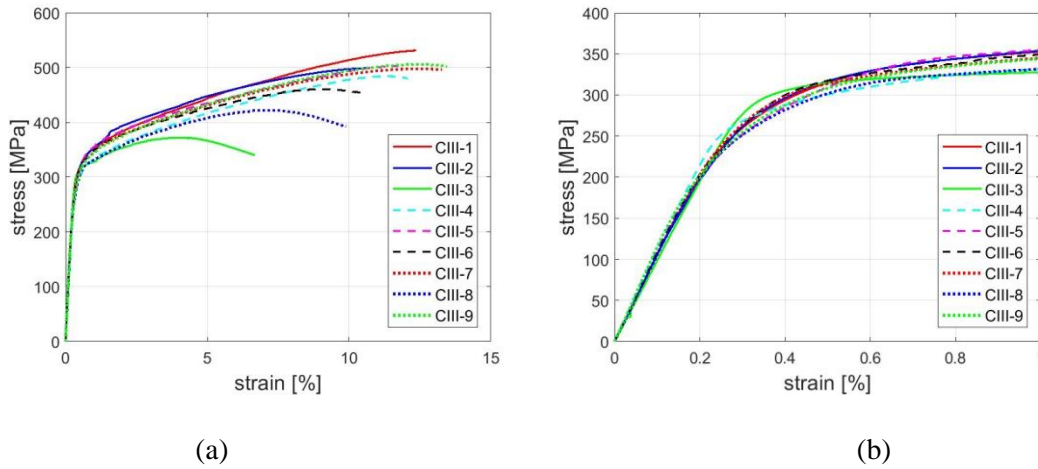
## 8.2. Test results

The stress-strain diagrams obtained from the tests of Series III are shown in graphs of Figure 21. The values of effective cross-sectional areas from volume measurements ( $A_{av,v}$ ) as described in Section 6.3 are considered to compute the stresses from the recorded values of forces. Values of strains are computed as the average between the values recorded by the two LVDTs at the two sides of the specimen. It has been verified that, up to the first yielding, the deformation values recorded by the two LVDTs remained close enough to each other so that the average deformation can be considered as representative of the axial strain. Table 5 collects the numerical values of the main mechanical parameters.

As first evaluation it can be noted that the elastic behavior in compression is similar to the one observed from the tensile tests, with Young's modulus values ( $E_0$ ) around 100 GPa, and 0.2% proof stress  $\sigma_{0.2}$  values around 300 MPa. The ultimate stress and strain values ( $\sigma_u$  and  $\epsilon_u$ ) are affected by the specific kind of local buckling failure and therefore can be used as a reference only for similar specimens (e.g. hollow circular specimens with diameter/thickness ratio equal to 10). Moreover, the compressive specimens tend to have a rounded stress-strain response, as visible from Figure 21b, similar to the behavior of "dog-bone" shaped specimens cut along the transversal direction (Series II), as visible from Figure 18f.

It can be of interest to notice that the local buckling mode of failure was highly influenced by the presence of the geometrical imperfection and typically resulted in an asymmetric deformed configuration (see Figure 20). Nonetheless, specimen CIII-1 exhibited a more symmetric mode of failure resulting in an overall increase in the peak stress (equal to roughly 530 MPa and closer to the

1 values obtained from the tensile tests) without the typical final drop evidenced by the stress-strain  
 2 curves of most of the specimens.



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5 *Figure 21: Stress-strain curve from compressive tests data (Series III): (a) full response; (b) zoom for strains < 1%.*

6  
7 *Table 5: Results from compressive tests (Series III).*

Specimen	$\sigma_{0.2}$ [MPa]	$\sigma_u$ [MPa]	$\epsilon_u$ [%]	$E_0$ [GPa]
CIII-1	317	531	12	106
CIII-2	318	499	11	95
CIII-3	317	372	7	98
CIII-4	293	484	12	102
CIII-5	303	502	12	106
CIII-6	317	461	11	103
CIII-7	308	498	13	104
CIII-8	294	422	10	104
CIII-9	299	506	13	115
$\mu$	<b>307</b>	<b>475</b>	<b>11</b>	<b>104</b>
COV	<b>0.03</b>	<b>0.10</b>	<b>0.16</b>	<b>0.05</b>

8  
9 **9. Discussion of the mechanical tests results**

10 Although the results presented in the previous sections are based on limited number of samples  
 11 (especially the ones related to smooth specimens of Series I), they provide first indications useful for  
 12 structural design purposes as well as to plan future experimental campaigns.

13 Tables 6 provides the values of key mechanical parameters obtained from of the experimental tests  
 14 with reference to smoothed and rough surfaces, tested along the parallel and perpendicular directions  
 15 with respect to the printing layer, both in tension and in compression.

1 The mechanical properties from tensile tests on artificially smoothed samples either in terms of  
 2 Young's modulus, 0.2% proof stress  $\sigma_{0.2}$  and ultimate stress  $\sigma_u$  seem to be practically coincident in  
 3 both longitudinal and orthogonal directions (with relative differences off less than 5%). This indicates  
 4 that the printing direction may not lead to an orthotropic behavior of the base material.

5 The mechanical properties from tensile tests on rough (as manufactured) samples indicates that  
 6 Young's modulus, 0.2% proof stress  $\sigma_{0.2}$  and ultimate stress  $\sigma_u$  along the transversal direction are  
 7 smaller than the corresponding values along the parallel directions (relative differences of 10-20%).  
 8 This indicates that the surface irregularities may lead to a geometrically-induced orthotropic behavior  
 9 of the structural members.

10 The mechanical properties from compression tests on rough (as manufactured) samples, relative to  
 11 the orthogonal direction only, indicate that both Young's modulus and 0.2% proof stress  $\sigma_{0.2}$  are  
 12 practically coincident to the corresponding ones from tensile tests. On the other hand, ultimate stress  
 13 in compression  $\sigma_u$  shows a reduction of about 10% with respect to the one from tensile test, due to  
 14 the specific buckling failure mode. This indicates that neither the geometrical irregularities nor the  
 15 thermal effects (and their combination) associated with WAAM manufacturing process induces  
 16 significant asymmetric tension-compression behavior (up to yielding) of the structural members.

17 *Table 6: Summary of the experimental test results .*

	Tensile					Compression (rough, $\perp$ )
	Smoothed surface		Rough surface			
	$\parallel$	$\perp$	$\parallel$	$\perp$	Relative difference	
$E_0$ [MPa]	113	108	128	104	-19%	104
$\sigma_{0.2}$ [MPa]	353	355	347	303	-13%	307
$\sigma_u$ [MPa]	554	535	569	521	-8%	475

18

## 19 **Conclusions**

20 In the paper, the results of a first experimental campaign aimed at characterizing the geometrical and  
 21 mechanical properties of Wire-and-Arc Additive Manufactured (WAAM) stainless steel structural  
 22 members are presented. The main goal of this first work is to draw some preliminary considerations  
 23 useful for structural design and planning future exhaustive experiment campaigns.

24 Firstly, the attention has been paid on the geometrical characterization. Results obtained from manual  
 25 measurements, volume-based measurements and 3D scanning acquisition system have been analyzed  
 26 and compared. In details, the effort has been focused in (i) evaluating the discrepancies between the

1 nominal and the effective thickness of the specimens extracted from the planar plates which have  
2 been quantified, on average, in the order of  $\pm 0.5$  mm ( $\pm 10\%$  of the nominal value), (ii) evaluate the  
3 discrepancies between the nominal and the effective cross-section outer diameter of hollow tubes,  
4 quantified in the order of  $-2$  mm ( $-25\%$  of the nominal value) and (iii) evaluate the lack of straightness  
5 of hollow tubes which can be quantified, at first approximation, as an initial imperfection with  
6 maximum value of about  $1/300$  of the member length.

7 Secondly, the main mechanical parameters of WAAM stainless steel material have been investigated  
8 through tensile tests on artificially polished samples where all geometrical imperfections were  
9 removed. The most relevant outcome from a structural design point of view is the resulting low value  
10 of Young's modulus for WAAM elements, in the order of 110-130 GPa, highly reduced with respect  
11 to traditionally manufactured stainless steel material. Nonetheless, values of 0.2% proof stresses as  
12 well as ultimate stresses are in line with those of traditionally manufactured stainless steel material.  
13 Also, the obtained results indicate that the mechanical behavior of the basic material is not affected  
14 by the orientation of the specimen with respect to the printing direction.

15 Finally, the mechanical properties of as-manufactured structural specimens have been investigated  
16 through tensile and compression tests. In order to account for the geometric imperfections, a  
17 simplified approach has been introduced to interpret the experimental results based on the concept of  
18 effective cross-sectional area which could be useful also for design purposes. The obtained results  
19 indicate that the mechanical behavior of as-manufactured pieces is affected (through the geometrical  
20 imperfections) by the orientation of specimen with respect to the printing direction. Indeed, the  
21 specimens cut perpendicular to the direction of printing layer (e.g. transversal direction) have reduced  
22 strengths with respect to those cut along the printing direction (e.g. longitudinal direction). In details,  
23 0.2% proof stress and ultimate stress along the transversal direction are respectively reduced of about  
24 10% with respect to those along the longitudinal direction. Also Young's modulus in the transversal  
25 direction is reduced of about 20% with respect to the one along the longitudinal direction. Moreover,  
26 no evident asymmetric behavior in tension and compression has been observed up to yielding.

27 Overall, the tests indicate a marked orthotropic behavior of WAAM members with geometrical  
28 irregularities due to the manufacturing process. On the other hand, the WAAM material seems not  
29 exhibiting an orthotropic behavior.

30

## 31 **Acknowledgements**

1 The support of Dutch company MX3D, which provided the tested materials, is gratefully  
2 acknowledged. The two anonymous reviewers are gratefully acknowledged. We believe that their  
3 comments and suggestions significantly contributed to improve the quality of the paper.

4

## 5 **Notation list**

6 *The following symbols are used in this paper:*

7  $A_{av,c}$  = average cross-sectional area from caliper hand measurements;

8  $A_{av,v}$  = average cross-sectional area from volume-based measurement;

9  $A_{eff}$  = effective cross-sectional area;

10  $A_{real}$  = effective cross-sectional area from printed specimen;

11  $D_1$  = first dimension of the circumscribed polygon;

12  $D_2$  = second dimension of the circumscribed polygon;

13  $D_{av,c}$  = average outer diameter from caliper hand measurements;

14  $D_{av,s}$  = average outer diameter from 3D scan measurements;

15  $D_c$  = outer diameter from caliper hand measurement;

16  $D_n$  = outer diameter from digital model;

17  $D_{real}$  = real outer diameter from printed element;

18  $D_s$  = outer diameter from 3D scan measurement;

19  $E_0$  = Young's modulus;

20  $E_{0,r\parallel}$  = Young's modulus values from rough specimens cut along the printing layer;

21  $E_{0,r\perp}$  = Young's modulus values from rough specimens cut perpendicular to the printing layer;

22  $E_{0,s\parallel}$  = Young's modulus values from artificially smoothed tensile specimens cut along the printing  
23 layer;

24  $E_{0,s\perp}$  = Young's modulus values from artificially smoothed tensile specimens cut perpendicular to  
25 the printing layer;

26  $F$  = axial force applied to the specimen;

- 1  $L_{av,c}$  = average specimen length from caliper hand measurement;
- 2  $L_c$  = specimen length from caliper hand measurement;
- 3  $L_n$  = specimen length from digital model;
- 4  $L_{real}$  = real specimen length from printed element;
- 5  $O$  = out-of-roundness;
- 6  $V$  = volume measure;
- 7  $c_{real}$  = position of the cross-sectional centroid taken from 3D scan measurements;
- 8  $c_{real,r}$  = radial coordinate of the position of the cross-sectional centroid taken from 3D scan
- 9 measurements;
- 10  $c_{real,x}$  = x-coordinate of the position of the cross-sectional centroid taken from 3D scan measurements;
- 11  $c_{real,y}$  = y-coordinate of the position of the cross-sectional centroid taken from 3D scan measurements;
- 12  $m_{air}$  = specimen mass measured in air;
- 13  $m_{H_2O}$  = specimen mass measured inside water;
- 14  $n$  = Ramberg-Osgood parameter;
- 15  $r$  = polar coordinate corresponding to the value of radius along the tubular specimen;
- 16  $s_{av,c}$  = average thickness of circular hollow specimens from caliper hand measurements;
- 17  $s_{av,v}$  = average thickness of circular hollow specimens from volume-based measurements;
- 18  $s_c$  = thickness of circular hollow specimens from caliper hand measurement;
- 19  $s_n$  = nominal thickness of circular hollow specimens from digital model;
- 20  $s_{real}$  = effective thickness of circular hollow specimens from printed element;
- 21  $t_{av,c}$  = average thickness of “dog-bone” specimens from caliper hand measurements;
- 22  $t_{av,v}$  = average thickness of “dog-bone” specimens from volume-based measurements;
- 23  $t_c$  = thickness of “dog-bone” specimens from caliper hand measurement;
- 24  $t_n$  = nominal thickness of the plate from digital model;

- 1  $t_{\text{real}}$  = real thickness of the plate from printed element;
- 2  $\varepsilon$  = value of strain (for Ramberg-Osgood equation);
- 3  $\varepsilon_u$  = ultimate strain (at rupture);
- 4  $\gamma_{\text{H}_2\text{O}}$  = average density of water;
- 5  $\gamma_v$  = average density of specimen from volume measurement;
- 6  $\mu$  = mean value from distributions;
- 7  $\mu_{D_s}$  = mean value from distribution of outer diameter values taken from 3D scan at 40 cross sections;
- 8  $\sigma$  = value of stress (for Ramberg-Osgood equation);
- 9  $\sigma_{D_s}$  = standard deviation of distribution of outer diameter values taken from 3D scan at 40 cross
- 10 sections;
- 11  $\sigma_{\text{eff}}$  = effective axial stress;
- 12  $\sigma_{0.01}$  = 0.01% proof stress;
- 13  $\sigma_{0.2}$  = 0.2% proof stress;
- 14  $\sigma_{0.2,r\parallel}$  = 0.2% proof stress values from rough specimens cut along the printing layer;
- 15  $\sigma_{0.2,r\perp}$  = 0.2% proof stress values cut from rough specimens cut perpendicular to the printing layer;
- 16  $\sigma_{0.2,s\parallel}$  = 0.2% proof stress values from artificially smoothed tensile specimens cut along the printing
- 17 layer;
- 18  $\sigma_{0.2,s\perp}$  = 0.2% proof stress values from artificially smoothed tensile specimens cut perpendicular to
- 19 the printing layer;
- 20  $\sigma_u$  = ultimate stress;
- 21  $\sigma_{u,r\parallel}$  = ultimate stress values from rough specimens cut along the printing layer;
- 22  $\sigma_{u,r\perp}$  = ultimate stress values cut from rough specimens cut perpendicular to the printing layer;
- 23  $\sigma_{u,s\parallel}$  = ultimate stress values from artificially smoothed tensile specimens cut along the printing layer;
- 24  $\sigma_{u,s\perp}$  = ultimate stress values from artificially smoothed tensile specimens cut perpendicular to the
- 25 printing layer;
- 26  $\varphi$  = correction factor for cross-sectional area ( $t_{\text{av},v}/t_n$ );
- 27  $\varphi_{\text{av}}$  = average value of correction factor for cross-sectional area;



1  $\chi$  = ratio between thickness measures taken from caliper and thickness taken from volume measures  
2 ( $t_{av,c}/t_{av,v}$ ).

3  $\chi_{av}$  = average value of  $\chi$ .

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# Appendix A

Table A1: Hand measures of thickness of “dog-bone” specimens (Series I and II).

Specimen	Quantities based on caliper measures	
	$t_{av,c}$ [mm]	$A_{av,c}$ [mm]
TI-1-y	2.88	30.76
TI-1-x	2.95	29.39
TII-1-x	4.47	88.44
TII-2-x	4.67	93.33
TII-3-x	4.63	93.13
TII-4-x	4.13	82.67
TII-1-y	4.33	86.67
TII-2-y	4.33	90.13
TII-3-y	4.13	84.32
TII-4-y	4.30	85.57
$\mu$	<b>4.38</b>	<b>88.03</b>
COV	<b>0.04</b>	<b>0.04</b>

Table A2: Hand and volume-based measures of thickness of “dog-bone” specimens (Series VI).

Specimen	Quantities based on caliper measures		Quantities based on volume measures			
	$t_{av,c}$ [mm]	$A_{av,c}$ [mm <sup>2</sup> ]	$\gamma_v$ [g/cm <sup>3</sup> ]	$t_{av,v}$ [mm]	$A_{av,v}$ [mm <sup>2</sup> ]	$\chi$ [-]
TVI-1-x	4.47	85.14	7.71	3.56	70.44	1.26
TVI-2-x	4.67	96.00	8.12	3.32	66.33	1.41
TVI-3-x	4.33	84.00	7.79	3.66	73.19	1.18
TVI-4-x	4.33	85.28	7.99	3.55	73.74	1.22
TVI-5-x	4.17	82.00	7.93	3.59	71.88	1.16
TVI-6-x	4.03	75.48	7.84	3.52	71.77	1.14
TVI-7-x	4.30	82.56	7.87	3.79	72.81	1.13
TVI-8-x	4.37	84.00	7.89	3.66	73.18	1.19
TVI-9-x	4.47	118.68	7.99	3.42	94.26	1.31
TVI-10-x	4.43	121.00	7.94	3.66	100.71	1.21
TVI-11-x	4.70	122.40	7.99	3.75	95.74	1.25
TVI-12-x	4.53	129.25	7.99	3.58	98.56	1.27
TVI-13-x	4.93	135.36	7.98	4.50	126.87	1.10
TVI-14-x	4.93	106.22	7.97	4.37	98.67	1.13
TVI-15-x	4.87	110.74	8.02	4.64	104.81	1.05
TVI-16-x	4.60	124.20	7.80	4.13	113.98	1.11
TVI-17-x	4.97	135.24	7.92	4.32	119.21	1.15
TVI-18-x	4.77	141.00	7.99	4.23	126.86	1.13
TVI-19-x	4.93	135.24	7.93	4.26	117.46	1.16
TVI-20-x	4.93	123.97	8.01	4.30	108.82	1.15

TVI-1-y	4.63	98.49	7.37	4.00	80.31	1.16
TVI-2-y	4.13	84.00	7.35	3.99	79.82	1.04
TVI-3-y	4.13	81.60	7.89	3.66	74.59	1.13
TVI-4-y	4.30	91.54	8.04	3.62	72.01	1.19
TVI-5-y	4.10	79.95	7.95	3.62	70.63	1.13
TVI-6-y	4.17	82.32	7.86	3.71	72.74	1.12
TVI-7-y	4.20	84.00	7.78	3.74	74.89	1.12
TVI-8-y	4.33	84.84	7.82	3.72	75.18	1.16
TVI-9-y	4.10	105.20	8.66	3.30	86.90	1.24
TVI-10-y	4.10	107.01	8.72	3.36	87.79	1.22
TVI-11-y	4.10	113.16	7.90	3.69	101.82	1.11
TVI-12-y	4.17	104.14	7.99	3.57	90.79	1.17
TVI-13-y	4.83	107.16	7.95	4.33	98.63	1.12
TVI-14-y	4.67	111.23	7.97	4.29	97.38	1.09
TVI-15-y	4.80	133.00	7.48	4.56	121.40	1.05
TVI-16-y	4.90	130.00	7.72	4.43	115.14	1.11
TVI-17-y	5.03	149.94	7.92	4.40	129.40	1.14
TVI-18-y	5.07	155.55	7.98	4.47	136.43	1.13
TVI-19-y	4.70	121.92	7.98	4.24	107.78	1.11
TVI-20-y	4.63	129.25	7.99	4.23	116.42	1.09
<b>μ</b>	<b>4.52</b>	<b>108.30</b>	<b>7.93</b>	<b>3.92</b>	<b>94.23</b>	<b>1.16</b>
<b>COV</b>	<b>0.07</b>	<b>0.20</b>	<b>0.03</b>	<b>0.10</b>	<b>0.21</b>	<b>0.06</b>

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Table A3: measures of cross-sectional area for stub columns (Series III).

Specimen	Quantities based on caliper measurements			$L_{av,c}$ [mm]	Quantities based on volume measurements			$\chi$ [-]
	$D_{av,c}$ [mm]	$S_{av,c}$ [mm]	$A_{av,c}$ [mm <sup>2</sup> ]		$\gamma_v$ [g/cm <sup>3</sup> ]	$A_{av,v}$ [mm <sup>2</sup> ]	$S_{av,v}$ [mm]	
CIII-1	53.59	5.46	825.52	151.63	7.88	609.32	3.90	1.40
CIII-2	54.01	5.45	831.71	149.43	7.87	612.42	3.89	1.40
CIII -3	54.25	5.40	828.78	149.25	7.82	560.65	3.52	1.53
CIII -4	54.22	4.95	765.84	148.19	7.83	598.81	3.78	1.31
CIII -5	53.98	5.31	811.56	152.39	7.82	607.27	3.86	1.38
CIII -6	54.06	5.32	814.26	151.20	7.83	607.15	3.85	1.38
CIII -7	53.83	4.97	762.83	150.97	7.83	609.26	3.88	1.28
CIII -8	54.19	5.46	835.13	151.97	7.82	603.23	3.81	1.43
CIII -9	53.65	4.94	755.61	152.54	7.83	601.61	3.84	1.29
<b>μ</b>	<b>53.97</b>	<b>5.25</b>	<b>803.47</b>	<b>150.84</b>	<b>7.84</b>	<b>601.08</b>	<b>3.81</b>	<b>1.38</b>
<b>COV</b>	<b>0.01</b>	<b>0.04</b>	<b>0.04</b>	<b>0.01</b>	<b>0.01</b>	<b>0.03</b>	<b>0.03</b>	<b>0.06</b>

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Table A4: 3D-scan measures of BIV-3 tubular element over 40 cross-sections.

<b>Cross-section height [mm]</b>	<b>D<sub>s</sub> [mm]</b>	<b>C<sub>real,x</sub> [mm]</b>	<b>C<sub>real,y</sub> [mm]</b>	<b>  C<sub>real,r</sub>   [mm]</b>
20	52.95	-1.03	-0.51	1.15
40	52.95	-0.72	-0.40	0.82
60	52.17	-0.54	-0.46	0.71
80	53.07	-0.63	-0.77	0.99
100	53.37	-1.12	0.11	1.12
120	53.13	-0.59	-0.34	0.68
140	53.92	-0.05	-0.47	0.47
160	53.65	-0.64	-0.52	0.82
180	54.54	0.13	-0.80	0.81
200	53.49	0.19	-0.94	0.95
220	53.96	0.51	0.20	0.55
240	53.50	0.23	0.38	0.44
260	53.14	-0.01	0.24	0.24
280	53.62	0.22	0.28	0.44
300	53.51	-0.53	-0.07	0.54
320	53.77	-0.61	0.18	0.63
340	53.68	-0.76	-0.14	0.77
360	53.68	-0.37	-0.70	0.79
380	53.54	-0.41	-0.52	0.66
400	53.24	-0.17	0.34	0.39
420	54.00	0.01	-0.06	0.06
440	52.98	0.00	-0.13	0.13
460	53.23	-0.71	-0.60	0.93
480	53.62	-0.65	-1.44	1.58
500	53.70	0.19	-0.77	0.80
520	53.21	-0.20	-0.73	0.76
540	53.87	-0.08	-0.25	0.26
560	53.77	-0.85	-0.25	0.89
580	53.59	-1.27	-0.40	1.33
600	53.12	-2.39	-0.82	2.52
620	53.61	-0.89	-1.14	1.45
640	53.32	-1.57	-0.76	1.74
660	53.76	-0.79	-1.16	1.41
680	52.76	-0.79	-1.16	1.41
700	53.64	-1.05	-0.44	1.14
720	53.16	-0.39	-0.16	0.42
740	53.53	-0.24	-0.37	0.44
760	53.40	-0.20	-0.23	0.30
780	53.36	-0.08	0.18	0.20
800	53.60	0.41	0.31	0.51
<b>μ</b>	<b>53.48</b>	<b>-0.46</b>	<b>-0.38</b>	<b>0.81</b>
<b>COV</b>	<b>0.01</b>	<b>1.23</b>	<b>1.22</b>	<b>0.61</b>